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LIVERMORE
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LLNL-SR-616753

Structural Upgrade Study Building 313

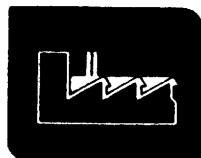
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February 7, 2013

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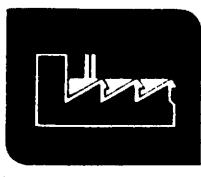
OAKLAND

LONG BEACH

STRUCTURAL UPGRADE STUDY
BUILDING 313
UNIVERSITY OF CALIFORNIA
LAWRENCE LIVERMORE NATIONAL LABORATORY

Livermore, California

Prepared for
University of California
Lawrence Livermore National Laboratory
Livermore, California
by
Frederiksen Engineering Co., Inc.
July 17, 1981



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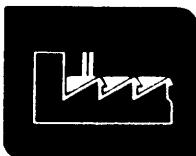
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STRUCTURAL REPORT BUILDING 313

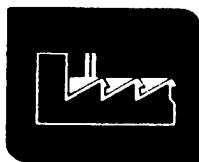
INTRODUCTION

The intent of this study is to identify and establish whether or not the present structural design of Building 313 meets the design requirements found within this report. The building evaluation includes dynamic forces due to seismic, wind and missile. Structural deficiencies are pointed out along with a recommended course of corrective action for repair or upgrade, with an approximate estimate of construction cost for each. Drawings, specifications and calculations are provided as part of the recommended course of corrective action.

DESCRIPTION OF STRUCTURE

Building 313, "LLNL Disaster Center" is located at Lawrence Livermore National Laboratory, Livermore, California. This building is a single story structure which measures 75'-6" long x 40'-5" wide x 12'-0" high, with a basement 27'-0" x 40'-5" x 9'-0", located close to the middle of the building. The roof framing is a metal decking on steel beams, supported by eight inches (8") of reinforced concrete walls (see Dwg. 1354-A1, A2). The slab on grade measures six inches (6") thick, and the slab over the basement is fourteen inches (14") thick with top and bottom reinforcement at both directions.

The lateral force resisting system for this building is a box-typed shear wall. Lateral forces, whether they be statically applied or dynamically induced, are transmitted by the metal roof diaphragm to the shear walls and the foundations.



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DESIGN CRITERIA

Building 313 was designed and built in the sixties when seismic design criteria was low. Our objective is to evaluate and upgrade the structure to withstand the following requirements.

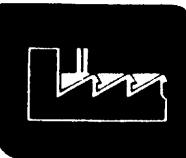
1. Seismic

Two levels of seismic loads are considered.

a. Using the LLNL Ground Response Spectra with a corresponding horizontal ground acceleration of 0.5g. Engineering evaluation and design are accomplished by using Uniform Building Code Analysis Method along with UBC Strength Allowables. Connection evaluation and design are accounted for an additional load factor of 1.5.

b. Using LLNL Ground Response Spectra with a corresponding horizontal ground acceleration of 0.8g. Engineering evaluation and design are accomplished by inelastic analysis method, as to insure prevention of collapse and to allow continued operation of Building 313 following an earthquake having major intensity at the site.

The maximum vertical ground acceleration shall be equal to two-thirds (2/3) of the maximum horizontal ground acceleration. The horizontal ground motion shall be considered to act simultaneously with the vertical ground motion (two-directional excitation).



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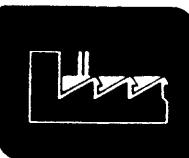
STRUCTURAL REPORT BUILDING 313

2. Wind

Building 313 has to upgrade to withstand wind pressure of 110 mph or (30 psf).

3. Missile

Building 313 has to upgrade to withstand missile impact such as a 2" x 4" - 12 foot long timber, weighing 20 pounds and traveling at 70 mph.



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ANALYSIS AND DESIGN

1. Seismic

A simplified three dimensional model has been set up for the dynamic analysis. (See Fig. 1) This model is assumed fixed at the base with the concrete walls replaced by the shell elements and the metal roof diaphragm replaced by the membrane elements and beams. The computer program "EASE 2" has been used for the dynamic analysis to determine the building story drift, out of plane flexure of the wall, and the shear stress of the metal roof diaphragm, with respect to the seismic level of 0.5g and 0.8g accordingly. Assume there is little cracking at the walls, four (4) percent damping is used throughout the analysis.

From the acceleration response spectra curves (See Fig. 5) the acceleration stays constant at 1.0g as the natural frequency reaches 30 hertz per cycle. This means that a building with a natural frequency less than 30 hertz per cycle responds dynamically according to the response spectra at or below the frequency of 30 hertz per cycle, and beyond that range a building can be assumed to respond statically. So in general practice, eigenvalues and eigenvectors are generated to the frequency of 30 hertz per cycle, and the percentage of total mass participated within that frequency are obtained and used for dynamic response analysis, while the remaining percentage of total mass are used for static analysis.

The results of the analysis are combined to check for maximum stress. Theoretically for a better result of analysis, the number of eigenvalues and eigenvectors has to be generated until a high percentage of total mass participated (about 90%) is obtained before running the dynamic response analysis, but it may be costly to do so. Three mode shapes have been plotted as shown at Fig. 2, 3 & 4.



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The model shows no openings at the south wall, and it is planned to have several openings in the near future. In order to have the stiffness of this wall corrected for analysis, modulus of elasticity was adjusted to compensate for the future openings. Stresses obtained should not be far away from reality.

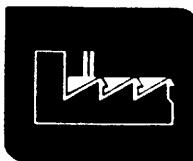
The basement wall is nine feet (9'-0") below finished grade. Horizontal and vertical dynamic soil pressure are calculated by the use of Mononobe-Okabe theory, and this has to combine with the static soil pressure for evaluating the strength of the wall.

2. Wind

Wind pressure is smaller than seismic and it is not considered in this report.

3. Missile

A wind-borne missile impact test at the roof diaphragm has been done by the Civil Engineering Department of the Lawrence Livermore National Laboratory. For test report see Appendix C.



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RESULTS AND RECOMMENDATIONS

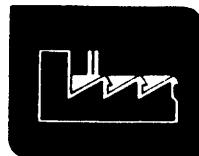
1. Seismic

Whereas Building 313 was not designed for the upgraded seismic design criteria being considered in this report, some structural deficiencies are to be expected.

From the dynamic response analysis, the weakest area is found at the metal roof diaphragm. Welded connection at the decking is not adequate for shear transfer to the east wall. Axial forces generated by seismic at the top of the wall caused buckling to the deck. In order to correct these deficiencies steel members are installed along the top of the wall together with a horizontal bracing system to replace the flexible metal diaphragm. (See Dwg. 1354-A2) With the new bracing system installed, out of plane movement at the wall is reduced and shows no overstress due to seismic forces.

Building 313 is a box-typed structure, where center of mass was located approximately at the middle of the bulding and about 8'-0" high from the ground. Overturning and sliding of the building is rarely to occur, and story drift is small as expected.

Basement wall is overstressed during seismic. A simple corrective way is to add metal plates at the inside face of wall at 5'-0" intervals. (See Dwg. 1354-A3) Shear stress is not significant at this wall. The retaining wall next to the steel stair at the north was also found overstressed. A new grade beam has to be provided on top, above the existing grade beam. (See Dwg. 1354-A4)



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BUILDING 313**

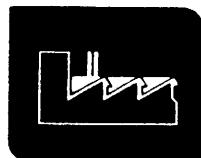
Future openings at the south wall shall be reinforced. (See Dwg. 1354-A6)
Out of plan forces will be transmitted through the steel columns to the
roof diaphragm and the foundation.

Slab over the basement:

The maximum live load capacity of this slab is 160 psf.

2. Missile

Building 313 is capable of withstanding the missile impact.



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COST ESTIMATE FOR STRENGTHENING

Basement Wall Reinforcing

1. Plate 3/8" x 12" x 8'-6" \$ 6,250.00

2. Concrete Grade Beam:

Demolish 5,000.00

Pour-in-place Concrete & Rebars 2,000.00

\$13,250.00

Total \$15,000.00*

New Roof Bracing System

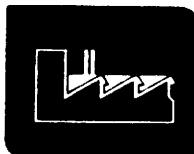
Bracing L 3x2x1 /4 \$12,000.00

C 7x9.8

Overhead & Profit **2,000.00**

Total \$14,000.00*

*Price includes materials and labor.



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APPENDIX A

Calculations

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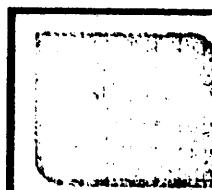
SUBJECT
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CALCULATION INDEX

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SHEET NO. 1 OF 12
JOB NO.

ROOF DIAPHRAGM

METAL DECKING SECT 3, 16 GAGE BY ROBERTSON.

SEISMIC C TRANS. DIRECTION 0.5g.

FROM COMPUTER OUTPUT,

SHEAR STRESS = (AVERAGE)

$$(1109 + 1002 + 977 + 997) \frac{1}{4} = 1021 \text{ psi}$$

SHEAR PER LINEAR FT C DIAPHRAGM =

$$N = 1021 \text{ psi} \times 0.0593" \times 12"$$

$$N = 732.7 \frac{\#}{\#}$$

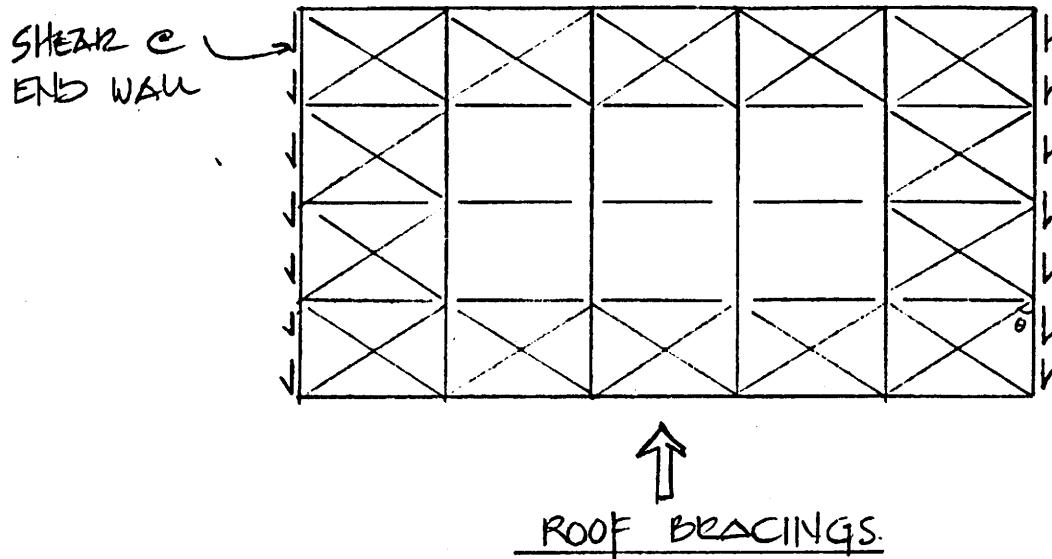
$$\text{REQ'D SHEAR } v' = 732.7 \frac{\#}{\#} \times 1.5 \\ = 1099 \frac{\#}{\#}$$

ALLOWABLE SHEAR = $600 \frac{\#}{\#} < 1099 \frac{\#}{\#}$, NG.

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CHKD. BY DATE

SUBJECT LLNL
BUDGET 3172

SHEET NO. 2 OF
JOB NO.



SINCE THE METAL ROOF DIAPHRAGM IS NOT ADEQUATE FOR HORIZ. SEISMIC FORCES, NEW BRACINGS ARE ADDED TO THE ROOF SYSTEM.

ASSUME BRACING MEMBERS TAKE TENSION ONLY

FOR SEISMIC C 0.5g.

THERE ARE 4 BRACING MEMBERS TO TAKE THE HORIZ. FORCE C EACH END WALL

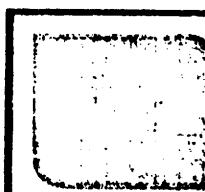
TOTAL END SHEAR C EA. WALL

$$V = 1099 \frac{1}{2} \times 40 \approx 44^k$$

$$\text{EACH BRACING MEMBER TAKES} = \frac{44^k}{4} = 11^k$$

$$\text{TENSION C EA. BRACING} = (\theta = \tan^{-1} \frac{16}{10} = 57.99^\circ)$$

$$T = \frac{11^k}{\cos 57.99} = 20.75^k \times 1.33 = 15^k$$



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TRY $\approx 3 \times 2 \times 4$ $l_y = 18.8'$, $l_x = 9.4'$

$$\frac{l}{l_x} = \frac{9.4 \times 12}{0.493} = 229 < 300 \text{ OK}$$

$$\frac{l}{l_y} = \frac{18.8 \times 12}{0.993} = 227 < 300 \text{ OK.}$$

TENSION CAPACITY =

$$21.67 \text{ kips} (1.19 - 0.75 \times 0.25) = 21.7 \text{ kips. } > 15'' \text{ OK}$$

USE - 2- $\frac{3}{4}'' \phi$ A 325 X BOLT

CAPACITY OF 2- $\frac{3}{4}'' \phi$ A325X BOLT

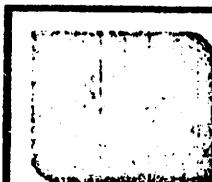
$$\text{SHEAR} = 13.3'' \times 2 = 26.6'' > 20.75'' \times 1.5 \times \frac{1}{1.33} \\ = 23.4'' \text{ OK}$$

BEARING @ GUSSET $\frac{1}{2} \times \frac{3}{8}''$

$$\text{AT } \frac{1}{2} \times \frac{3}{8}'' \quad 24.5'' > 23.4'' \text{ OK}$$

$$\text{AT } L3 \times 2 \times 4 \quad 2 \times 16.3'' = 32.6'' > 23.4'' \text{ OK}$$

USE	$L3 \times 2 \times 4$ w/ 2- $\frac{3}{4}'' \phi$ A325X BOLTS
-----	--



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ROOF DIAPHRAGM (CONT.)

SEISMIC CTEANS. DIRECTION = 0.8g.

SHEAR STRESS = (COMPUTER OUTPUT)

$$(1771 + 1602 + 1561 + 1602) \frac{1}{4} = 1634 \text{ psi}$$

SHEAR FORCE PER LINEAR FT

$$N = 1634 \times 0.059.8 \times 12 = 1173 \text{ #/ft}$$

TOTAL SHEAR @ EA. END WALL

$$V = 1173 \times 40'' \approx 47''$$

EACH BRACING MEMBERS TAKES $\frac{47}{4} = 11.75''$
say 12"

TENSION @ EA. BRACING

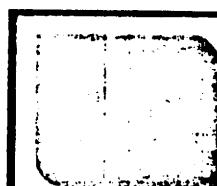
$$T = \frac{12}{\cos 57.99} \approx 23''$$

TENSION CAPACITY @ PLASTIC STRENGTH (L3x2x1/4)

$$= 36 \text{ ksi } (1.19 - 0.75 \times 0.25)$$

$$= 36'' \times 0.9 \\ \text{SAFE FACTOR}$$

$$= 32.5'' > 23'' \text{ S.K}$$



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EAST WALL $t = 8''$

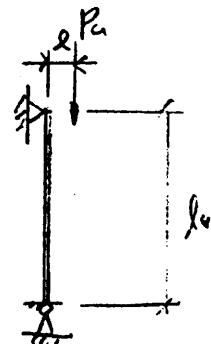
ROOF LOADING @ WALL

$$\begin{aligned}
 \text{D.L.} &= \left. \begin{array}{l} \text{ROOFING} = 7 \text{ PSF} \\ \text{DECKING} = 5 \text{ PSF} \\ \text{BEAM} = 1 \text{ PSF} \end{array} \right\} \times 13 \text{ PSF} \times 20' = 260 \text{ #/ft} \\
 \text{U.L.} &= 10 \text{ PSF} \times 20 \text{ ft} = 320 \text{ #/ft} \\
 \hline
 &\quad \overline{\qquad\qquad\qquad} \qquad\qquad\qquad \overline{\qquad\qquad\qquad} \\
 &= 580 \text{ #/ft}
 \end{aligned}$$

$$\text{WALL DL} = 100 \text{ psf} \times 12' = 1200 \text{ #/l}$$

VERT. FACTOR LOAD C WALL:

$$\begin{aligned} \text{D.L. } (260 + 1200) \times 1.4 &= 2.04\% \\ \text{L.L. } 320 \times 1.7 &= 0.54\% \end{aligned} \quad \left. \begin{array}{l} \text{D.L.} \\ \text{L.L.} \end{array} \right\} = 2.58\%$$



$$\text{MIN. ECCENTRICITY} = 0.1 \left(\frac{g}{12} \right) = 0.0677'$$

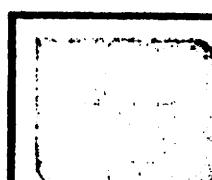
$$My_e = 2.58^u \times 0.0677' = 0.17^{lu}$$

MOMENT DUE TO SEISMIC = $0.8g(x)$

M_S = 2101 "#/ // CFROM computer)

$$M_{u/s} = 2101 \times 1.7 \times 1.1 = 3.93^{IK}$$

$$\sum M_w = 0.17 + 3.93 = 4.1^{IK} \leftarrow \checkmark \quad \text{center of wall}$$



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CHECK WALL SLENDERNESS =

$$\frac{kl_u}{r} = \frac{1 \times (12 \times 12)}{0.3 \times 8} = 60 < 100.$$

APPROX. EVALUATION OF SLENDERNESS EFFECTS
BY SECTION 10.11 MAY BE USED.

CALC. MOMENT MAGNIFICATION =

$$EI = \frac{Ec I_g / 2.5}{1 + \beta_d} \quad \beta_d = 0.$$

$$Ec = 57400 \sqrt{3000} = 3.14 \times 10^6 \text{ psi}$$

$$I_g = 12 \frac{(8)^3}{12} = 512 \text{ in}^4$$

$$Ec I_g = 3.14 \times 10^6 \times 512 = 1.6 \times 10^9$$

$$P_c = \frac{\pi^2 EI}{(kl_u)^2} = \frac{\pi^2 \times 1.6 \times 10^9}{(12 \times 12)^2} = 765^u$$

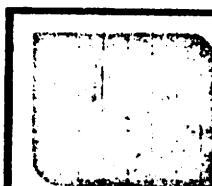
$$\delta = \frac{C_m}{1 - (\rho_u / \phi P_c)}$$

$$\phi = 0.9 - 0.2 (P_u / 0.1 f_c A_g)$$

$$= 0.9 - 0.2 (2.58 / 0.1 \times 3.14 \times 12 \times 8)$$

$$\phi = 0.88$$

$$\therefore \delta = \frac{1}{1 - (2.58 / 0.88 \times 765)} \approx 1.0$$



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$$\therefore M_u = 4.1 \times 1.0038 = 4.11^{\text{in}}$$

$$P_u = 2.58^{\text{in}}$$

COMBINE AXIAL FORCE & MOMENT:

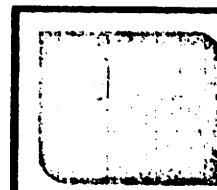
$$\begin{aligned} M_{us} &= M_u + N_u (d - \frac{h}{2}) \\ &= 4.11 + 2.58 (6.5 - \frac{8}{2}) \times \frac{1}{12} \\ &= 4.65^{\text{in}} \end{aligned}$$

$$F = \frac{12 \times 6.5^2}{12000} = 0.0423$$

$$K_u = \frac{4.65}{0.0423} = 110.$$

$$\alpha_u = 2.93$$

$$\begin{aligned} \text{REQ'D } A_s &= \frac{M_{us}}{\alpha_u d} - \frac{N_u}{\phi f_y} \\ &= \frac{4.65}{2.93 \times 6.5} - \frac{2.58}{0.85 \times 40} \\ &= 0.1683 \text{ in}^2 < 0.2 \text{ in}^2 \text{ O.K.} \end{aligned}$$



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INTERIOR WALL BETWEEN THE NEW & OLD BUILDING

VECT. LOAD

$$\text{Roof DL} = 13 \text{ psf} \times 14.5' = 188.5 \text{#/ft}$$

$$\text{Roof LL} = 16 \text{ psf} \times 14.5 = 232. \text{#/ft}$$

$$\text{WALL DL} 100 \text{ psf} \times 12' = 1200 \text{#/ft}$$

$$P_u = (188.5 + 1200) 1.4 + 232 \times 1.7 = 2.3 \text{#/ft}$$

MIN. ECCENTRIC MOMENT =

$$M_{ue} = 2.3 \times (0.33 + 0.065) = 0.91 \text{ in.}$$

$$M_{us} = 3.06 \text{ in.} \times 1.7 \times 1.1 \text{ (FROM computer output)} 0.89. \\ = 5.72 \text{ in.}$$

$$\Sigma M_u = 5.72 + 0.91 = 6.63 \text{ in.}$$

FROM CALC. ABOVE, IT SHOWS THAT THE MOMENT MAGNIFICATION FACTOR f IS VERY SMALL.

COMBINE AXIAL FORCE & MOMENT

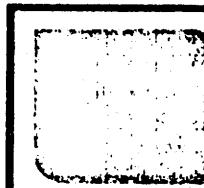
$$M_{us} = 6.63 + 2.3 (6.5 \times \frac{8}{2}) \frac{1}{12}$$

$$M_{us} = 7.11 \text{ in.}$$

$$F = \frac{12 \times 6.5}{12000} = 0.0423$$

$$K_u = \frac{7.11}{0.0423} = 160 \quad \alpha_u = 2.88$$

$$\text{REQ'D AS} = \frac{7.11}{2.88 \times 6.5} - \frac{2.3}{0.85 \times 40} = 0.31 \text{ in.}^2 \times 0.75 \\ = 0.22 \text{ in.}^2$$



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THE METAL DECKING (ROBECSON SECT 3, 16 GAUGE)
CANNOT TAKE MUCH OF THE AXIAL FORCES FROM
THE WALL DURING SEISMIC. STEEL BEAM IS DESIGN
TO TRANSFER HORIZ. LOADING FROM WALL TO THE NEW
BRACING DIAPHRAGM, THEN TO SHEAR WALL.

AXIAL LOADING @ THE METAL DECKING:

SEISMIC LEVEL @ 0.5g @ LONG. DIRECTION

AXIAL STRESS @ DECK

$$(213 + 585 + 594 + 251) \frac{1}{4} = 410.75 \text{#/in}^2$$

AXIAL FORCE =

$$410.75 \times 0.0598 \times 12 = 294.75 \text{#/ft.}$$

DESIGN STEEL BEAM TO TAKE LOADING OF 295#/ft.

$$M_{0.5g} = 0.295 \times \frac{10}{8} = 3.69 \text{IK}$$

SEISMIC @ 0.5g @ TRANS. DIRECTION

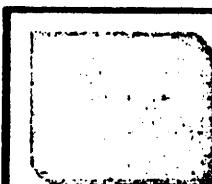
AXIAL STRESS @ DECK

$$(132 + 332 + 399 + 365 + 172) \frac{1}{5} = 280 \text{ psi}$$

AXIAL FORCE =

$$280 \times 0.0598 \times 12 = 201 \text{#/ft.}$$

$$M = 0.201 \times \frac{10}{8} = 6.43 \text{IK}$$



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SEISMIC @ LONG. DIRECTION = 0.8g. (X)

AXIAL STRESS @ DECK

$$(342 + 936 + 950 + 402) \frac{1}{4} = 657.5 \text{ psi}$$

AXIAL FORCE:

$$657.5 \times 0.0598 \times 12 = 472 \text{ #/ft}$$

$$M = 0.472 \times \frac{10^2}{8} = 5.9 \text{ ft-k}$$

SEISMIC @ TRANS. DIRECTION = 0.8g (z)

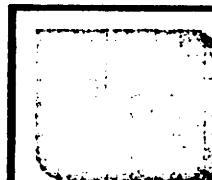
AXIAL STRESS: @ DECK

$$(210 + 532 + 637 + 583 + 274) \frac{1}{5} = 447.2 \text{ psi}$$

AXIAL FORCE

$$447.2 \times 0.0598 \times 12 = 321 \text{ #/ft}$$

$$M = 0.321 \times \frac{10^2}{8} = 10.27 \text{ ft-k}$$



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0.5g. C TENS. DIRECTION (z)

$$M = 6.43^k \quad (\text{ELASTIC})$$

$$\text{REQ'D } S = \frac{6.43 \times 12}{21.47} = 3.50 \text{ IN}^3$$

USE C X

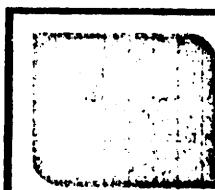
$$f_b = \frac{3.50 \times 12}{4.38} = 9.75 \text{ ksi OK}$$

$$\text{REACTION, R} = 201\% \times \frac{16}{2} = 1.6^k \times 1.5 \times 1.33 \\ = 1.8^k$$

CONNECTION = USE $\frac{3}{16}$ " WELD

4"- $\frac{3}{16}$ WELD GIVES

$$= \frac{3}{16} \times 0.707 \times 21.6 \times 4" = 11.45^k \text{ O.R.}$$



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SHEET NO. 12 OF _____
JOB NO. _____

Seismic @ girder

$$M_x = 133.4 \text{ "K} \quad (\text{strong axis})$$

$$F_y = 36 \text{ ksi}$$

$$M_y = 0.167 \text{ "K} \quad (\text{weak axis})$$

$$P = 2715 \text{ #}$$

$$\text{Static load} = 7 \text{ psf (Roofing)} + 5 \text{ psf (decking)} + 1 \text{ psf (beam)} = 13 \text{ psf}$$

$$\text{L.L.} = 20 \text{ psf (w/ Reduction, see Table 23-C, Page 139 UBC)} = 16 \text{ psf}$$

$$\text{Total load} = 29 \text{ psf}$$

$$\text{Span} = 15' - 9"$$

$$w = 29 \text{ psf} \times 15.75 \text{ ft} = 457 \text{ #/ft}$$

$$M = \frac{\omega l^2}{8} = \frac{457 \text{ #/ft} \times 40 \text{ ft}^2}{8} = 91.4 \text{ K'}$$

$$M_T = 133.4 \frac{\text{K''}}{12} + 91.4 \text{ K'} = 102.52 \text{ K'}$$

$$I = \frac{1}{12} b h^3 + A d^2 = \frac{1}{12} (6") (18)^3 - 2 \left[\frac{1}{12} (2.875 \times 17) \right] = 562 \text{ in}^4$$

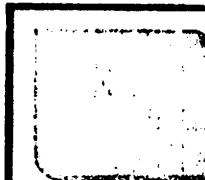
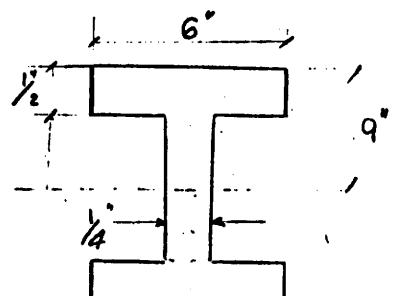
$$f_s = \frac{MC}{I} = \frac{(102.52)(12)(9)}{562 \text{ in}^4} = 19.7 \text{ ksi} < F_s = .6 F_y = 21.6 \text{ ksi OK}$$

$$f_v = \frac{VQ}{Ib}$$

$$Q = (3 \frac{\text{in}^2}{\text{in}^2} \times 8.75") + (2.125 \times 4.25") = 35.28 \text{ in}^3$$

$$f_v = \frac{\frac{1}{2}(457 \times 40)(35.28 \text{ in}^3)}{(562 \text{ in}^4)(\frac{1}{4})} = 2.3 \text{ ksi}$$

$$f_v = \frac{V}{dt_w} = \frac{\frac{1}{2}(457 \times 40)}{18" \times \frac{1}{4}"} = 2.08 \text{ ksi}$$

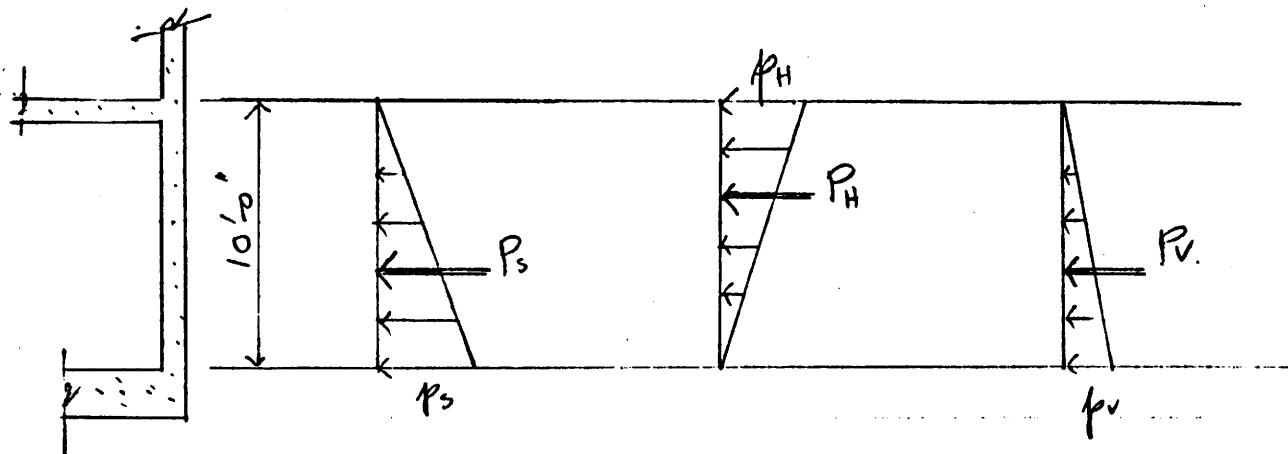


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BASEMENT WALL



A. STATIC SOIL
PRESSURE

B. HORIZ. DYN.
LAT. PRESSURE

C. VERT DYN
LAT. PRESSURE

A. STATIC SOIL PRESSURE : p_s

$$p_s = k_o \cdot \gamma \cdot H$$

$$p_s = \frac{1}{2} k_o \gamma H^2$$

B. HORIZ. DYNAMIC LATERAL PRESSURE , p_h (REF. MONONBE-KABEL THEORY)

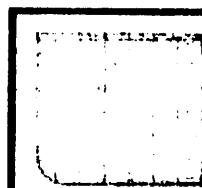
$$p_h = \gamma H (1 - 2v) K_a e$$

$$p_h = \frac{1}{2} \gamma H^2 (1 - 2v) K_a e$$

C. VERT. DYNAMIC LATERAL PRESSURE , p_v (REF. MONONBE-KABEL THEORY)

$$p_v = \gamma H K_a (\pm 2v)$$

$$p_v = \frac{1}{2} \gamma H^2 K_a (\pm 2v)$$



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SHEET NO. 14 OF
JOB NO. _____

WHERE

$$\gamma = 120 \text{ lb/ft}^3, H = 10 \text{ ft}$$

$$k_0 = 0.7$$

$$a_v = \text{VERTICAL GROUND ACCELERATION} = \frac{2}{3} a_H$$

$$a_H = \text{HORIZ. GROUND ACCELERATION.}$$

$$K_{AE} = \frac{\cos^2(\phi - \theta - \beta)}{\cos\theta \cos^2\beta \cos(\delta + \beta + \theta) \left[1 + \sqrt{\frac{\sin(\phi + \delta) \sin(\phi - \theta - i)}{\cos(\delta + \beta + \theta) \cos(i - \beta)}} \right]^2}$$

$$\theta = \tan^{-1} \frac{a_H}{1 - a_v}$$

$$\phi = \text{ANGLE OF FRICTION OF SOIL} = 30^\circ$$

$$\delta = \text{ANGLE OF WALL FRICTION} = 20^\circ$$

$$i = \text{slope of ground surface behind wall} = 0$$

$$\beta = \text{slope of back of wall to vertical} = 0$$

$$K_d = K_{AE} \text{ but set } \theta = 0$$

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SHEET NO. 15 OF _____
JOB NO. _____

CASE #1

$$\partial H = 0.5 g.$$

$$\partial V = \frac{2}{3} \times 0.5 g = 0.33 g.$$

A. STATIC SOIL FORCE, P_s .

$$P_s = 0.7 \times 120 \times 10 = 840 \text{#/in'}$$

$$P_s = \frac{1}{2} \times 10 \times 840 = 4200 \text{#/in'}$$

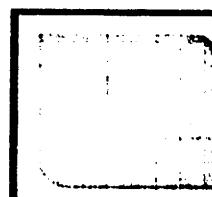
B. TOTAL DYN. LATERAL FORCE, P_H

$$\theta = \tan^{-1} \frac{0.5}{1 \pm 0.33} = 36.73^\circ \text{ OR } 20.60^\circ$$

$$K_{AE} = \frac{\cos^2(30 - 20.60)}{\cos 20.6 \cos(20 + 20.6) \left[1 + \frac{\sin(30 + 20) \sin(30 - 20.6)}{\cos(20 + 20.6)} \right]^2}$$
$$= 0.69$$

$$\therefore f_H = 120 \times 10 (1 - 0.33) 0.69 = 557.13 \text{#/in'}$$

$$P_H = \frac{1}{2} \times 10 \times 557.13 = \underline{2785.65 \text{#/in'}}$$



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C. VERT DYN. LATERAL FORCE , P_v

$$K_A = \frac{\cos^2 30}{\cos 20 [1 + \sqrt{\frac{\sin(30+20) \sin 30}{\cos 20}}]^2} = 0.2973$$

$$\therefore p_v = 120 \times 10 \times 0.2973 \times 0.33 = 117.74 \text{ kN}$$

$$P_v = \frac{1}{2} \times 10 \times 117.74 = 588.49 \text{ kN}$$

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SHEET NO. 17 OF _____
JOB NO. _____

CASE #2

HORIZ. GROUND ACCELERATION

$$a_H = 0.8 g.$$

VERT GROUND ACCELERATION

$$a_V = 0.8 g \times \frac{2}{3} = 0.533 g.$$

A. STATIC SOIL FORCE , P_s

$$P_s = 4200 \text{ kN}$$

B. HORIZ. DYN. LATERAL FORCE , P_H

ASSUME $\beta=0$, $i=0$ $\phi=30^\circ$ $\delta=20^\circ$

$$\theta = \tan^{-1} \frac{0.8}{1 \pm 0.533} = 27.56^\circ \text{ OR } 59.74^\circ$$

$$K_{AE} = \frac{\cos^2(30 - 27.56\phi)}{\cos 27.56\phi \cos^2 0 \cos(20 + 27.56) \left\{ 1 + \left[\frac{\sin(30 + 20) \sin(30 - 27.56)}{\cos(20 + 27.56) \cos 0} \right]^{1/2} \right\}^2}$$
$$= 1.1211$$

$$\therefore P_H = 120 \times 10 \times (1 - 0.533) 1.1211 = 027.8 \text{ kN}$$

$$P_H = \frac{1}{2} \times 10 \times 027.8 = 3139.0 \text{ kN}$$

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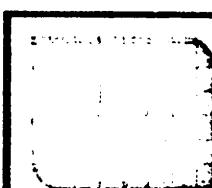
SHEET NO. 18 OF _____
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CASE #2 (CONT.)

c. VERT. DYN. LATERAL FORCE, P_v

$$p_v = 120 \times 10 \times 0.2973 (0.533) = 190.15 \frac{lb}{in}$$

$$P_v = \frac{1}{2} \times 10 \times 190.15 = 950.77 \frac{lb}{in}$$



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CASE #1 (0.5 g)

LOADING @ BASEMENT

$$U = 0.75(1.7H + 1.7 \times 1.1 E)$$

$$w_{uA} = 557.13 \times 1.7 \times 1.1 = 1042 \text{#/ft}$$

$$w_{uB} = (840 \times 1.7) + (117.74 \times 1.7 \times 1.1) = 1648 \text{#/ft}$$

$$R_A = 1042 \times \frac{10}{2} + 606 \times \frac{10}{2 \times 3} = 6220 \text{#}$$

$$R_B = 13450 - 6220 = 7230 \text{#}$$

ZERO SHEAR @ X = 20M R_A

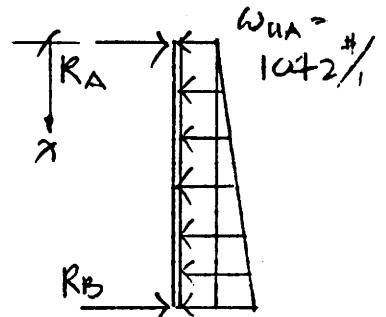
$$R_A - 1648x - \frac{60.6x^2}{2} = 0$$

$$6220 - 1648x - 30.3x^2 = 0$$

$$x = 3.54' \quad 5.2'$$

$$M_{u\max} = 6220 \times 3.54 - 1042 \times \frac{3.54^2}{2} - 214.73 \times \frac{3.54^2}{3 \times 2}$$
$$= 15021.86 \text{#} = 15 \text{ k}$$

$$0.75(15) = 11.25 \text{ in.}$$



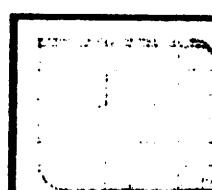
MOMENT CAPACITY:

$$f_c = 3000 \text{ psi}, d = 10'' - 1 = 9'' \quad A_s = 0.2 \text{ in.}^2$$

$$M_u = 3A_sd - 1.96A_s^2$$

$$= 3 \times 0.2 \times 9 - 1.96 \times 0.2^2$$

$$= 5.32 \text{ k} \times 1.35 = 7.08 \text{ k} \ll 15 \text{ k N.G.}$$



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SHEET NO. 20 OF _____
JOB NO. _____

CHECK SHEAR

$$V_u = 7230^{\dagger}$$

$$N = \frac{V_u}{\phi bd} = \frac{7230}{0.85 \times 12 \times 9} = 79 \text{ psi}$$

$$\text{ALLOWABLE SHEAR} = 2\sqrt{3000} = 109. \text{psi} > 79$$

O.K.

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JOB NO. _____

CASE #2 (0.8g)

$$U = 0.75 (1.7H + 1.7 \times 1.1E)$$

$$w_{uA} = 627.8 \times 1.7 \times 1.1 = 1174 \text{#/ft}$$

$$w_{uB} = 840.0 \times 1.7 + 190.15 \times 1.7 \times 1.1 = 1784 \text{#/ft}$$

$$R_A = 1174 \times \frac{10}{2} + 610 \times \frac{10}{2 \times 3} = 6887 \text{#}$$

$$R_B = 14790 - 6887 = 7903 \text{#}$$

ZERO SHEAR @ X FROM R_A

$$6887 - 1784x - \frac{61.0}{2}x^2 = 0$$

$$6887 - 1784x - 30.5x^2 = 0$$

$$x = 3.63'$$

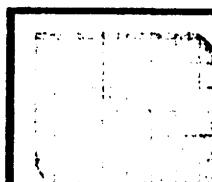
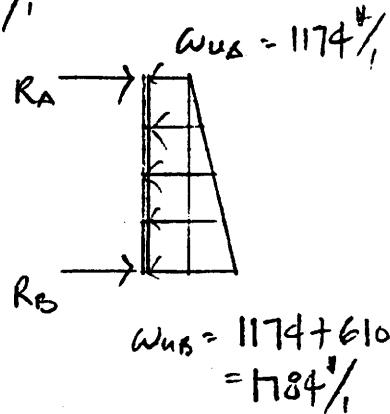
$$\begin{aligned} M_{u\max} &= 6887 \times 3.63 - 1174 \times \frac{3.63^2}{2} - 221.7 \times \frac{3.63}{3 \times 2} \\ &= 14788.3 \text{#} = \underline{\underline{18.8 \text{ k}}} \end{aligned}$$

MOMENT CAPACITY

$$M_u = 7.08 \text{ k} \ll 18.8 \text{ k} \quad \text{N.G. } \leftarrow$$

SHEAR

$$N = \frac{7903}{0.85 \times 12 \times 9} = 86 \text{ psi} < 109. \text{ psi o.k.}$$



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SUBJECT LLNL
BLDG. 313

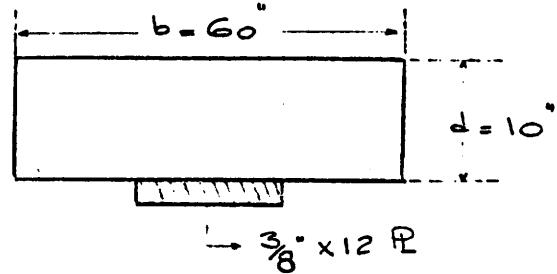
SHEET NO. 22 OF _____
JOB NO. _____

$$f'_c = 3000 \text{ Psi}$$

using $\frac{3}{8}'' \times 12''$ plates & 1" dia bolts

$$f_y = 36000 \text{ Psi}$$

$$A_s = (\frac{3}{8}'' \times 12'') - 2(1'' + \frac{1}{8}'')(\frac{3}{8}'') = 3.66 \text{ in}^2$$



$$A_s = Pbd$$

$$3.66 \text{ in}^2 = P \times 60'' \times 10''$$

$$P = 0.0061$$

$$m = \frac{f_y}{0.85 f'_c} = \frac{36000}{0.85 \times 3000} = 14.12$$

$$R_u = P f_y (1 - \frac{1}{2} \rho_m) = 0.0061 \times 36,000 (1 - \frac{1}{2} \times 0.0061 \times 14.12) = 210 \text{ Psi}$$

$$M_u = R_u b d^2 = 210 \text{ Psi} (60'') (10'')^2 (\frac{1}{12000}) = 105 \text{ K'}$$

$$M_u = \Phi M_u = .9 \times 105 \text{ K'} = 94.6 \text{ K'}$$

$$M_u = 94.6 \text{ K'} > 15 \text{ K'/ft} \times 5' = 75 \text{ K'} \quad \text{OK} \quad (\text{Case 1})$$

$$M_u = 94.6 \text{ K'} > 16.8 \text{ K'/ft} \times 5' = 84 \text{ K'} \quad \text{OK} \quad (\text{Case 2})$$

BY B.M. DATE 7/16

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SHEET NO. 23 OF

JOB NO.

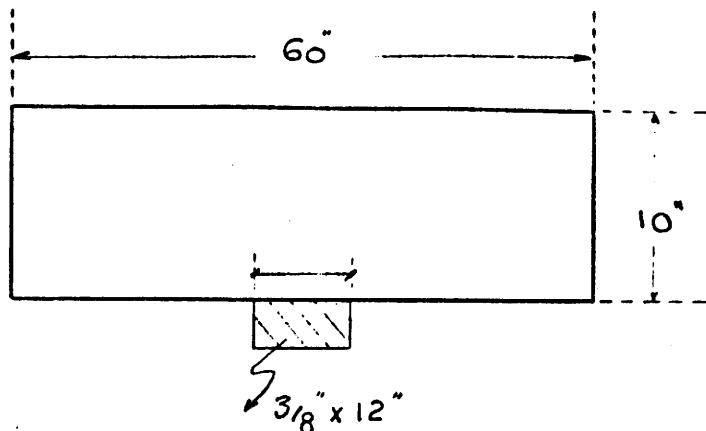
BLDG 313

$$f'_c = 3000 \text{ psi}$$

A-36 Steel $\Rightarrow f_y = 36000 \text{ psi}$

$$A_s = 4.5 \text{ in}^2$$

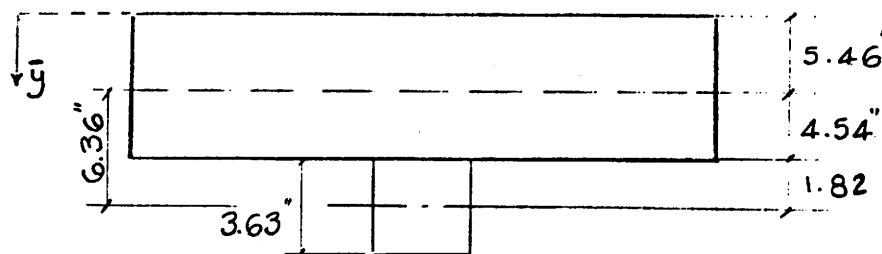
$$n = \frac{n_s}{n_c} = \frac{29}{3} = 9.67$$



Transformed Area :

$$A = 12'' \times (3\frac{1}{8}'' \times 9.67)$$

$$= 12'' \times 3.63 = 43.56 \text{ in}^2$$



$$\bar{y} = \frac{(60 \times 10 \times 5) + 43.56 (10 + \frac{3.63}{2})}{(10 \times 60) + (3.63 \times 12)} = 5.46''$$

$$I = \frac{1}{12} (60 \times 10^3) + (60 \times 10)(-46)^2 + \frac{1}{12} (12 \times 3.63^3) + 43.56 (6.36)^2$$

$$= 6,937 \text{ in}^4$$

$$Q = A \cdot y = (12'' \times 3.63'') (6.36'') = 277 \text{ in}^3$$

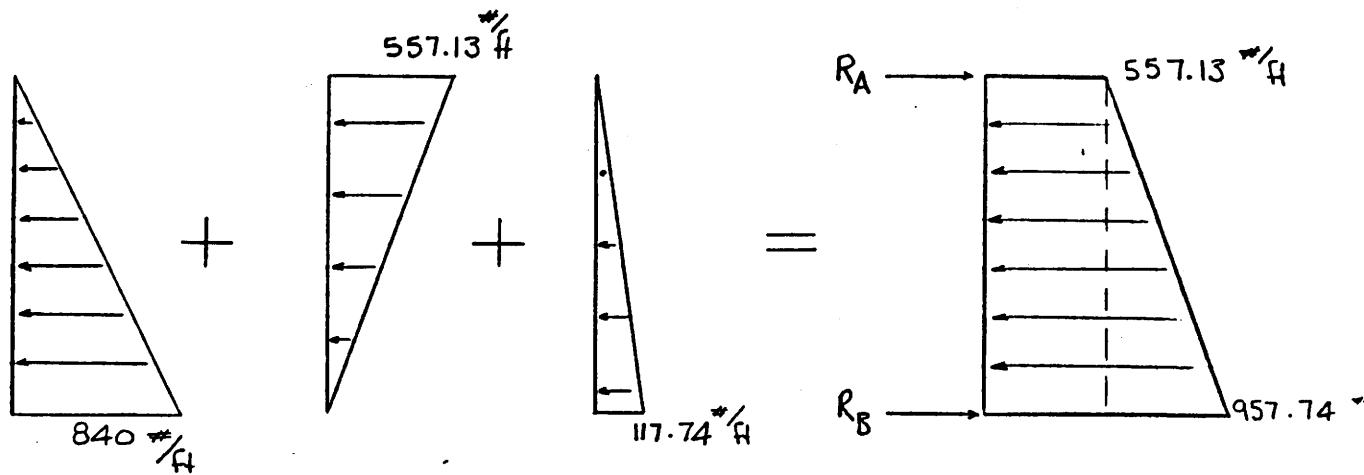
$$f_v = \frac{\sqrt{Q}}{I} = \frac{20600'' (277 \text{ in}^3)}{6937 \text{ in}^4} = 822.6 \text{ psi} \times 12 \times \frac{1}{1000} = 9.87 \text{ k/ft}$$

$$f_v = 9.87 \times 1.5 = 14.81 \text{ k/ft}$$

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BLDG. 313

SHEET NO. 24 OF
JOB NO.



$$R_A = \frac{1}{2}(557.13 \times 10) + \frac{400.16 \times 10}{2 \times 3} - 3453 \text{#/ft}$$

$$R_B = (557.13 \times 10) + (400.16 \times 10 \times \frac{1}{2}) - 3453 = 4120 \text{#/ft}$$

$$V = 4120 \text{#/ft} (5 \text{ft}) = 20600 \text{#}$$

$$f_v = \frac{20600^*(277\text{in}^3)}{6937 \text{ in}^4} = 822.6 \text{#/in} \times 12 \times \frac{1}{1000} = 9.87 \text{ k/in} \times 1.5 = 14.81$$

BY B.M. DATE 7/17
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SUBJECT LL.NL
BLDG. 313

SHEET NO. 25 OF
JOB NO. _____

Bolts:

Use "PHILLIPS RED HEAD" 7/8" Ø

$$\text{Shear Capacity} = 27000 \text{ #/bolt}, \text{ with a safe working load of } 25\% \\ = 27000 \times \frac{1}{4} \times \frac{1}{1000} = 6.75 \text{ k/bolt}$$

Seismic factor = 1.33

$$\text{Shear Capacity} = 1.33 \times 6.75 = 8.98 \text{ k/bolt}$$

$$\text{Spacing} = \frac{8.98 \text{ k}}{14.81 \text{ k/ft}} = 0.61 \text{ bolt/ft} \times 12 = 7.28 \text{ bolt/in}$$

USE 6" Spacing

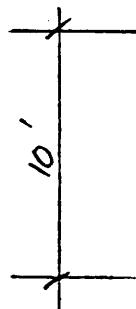
BY R.CHAN DATE 7/81
CHKD. BY _____ DATE _____

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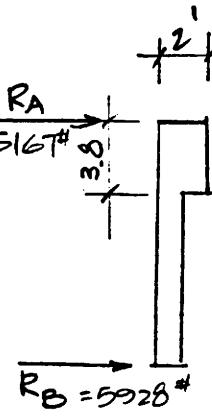
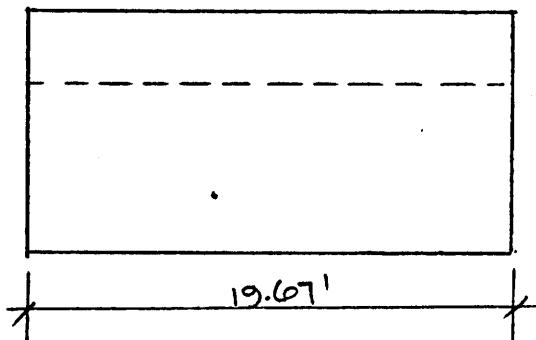
SHEET NO. 20 OF _____
JOB NO. _____

CASE 2. HORIZ. GROUND ACCELERATION (0.8g) (CONTROL)

LOADING @ WALL



(WALL) ELEV.



$$G\omega_{uA} = 881 \text{#/in}$$

$$\begin{aligned} G\omega_{uB} &= 881 + 457 \\ &= 1338 \text{#/in} \end{aligned}$$

SECTION.

$$U = 0.75 (1.7 p_s + 1.7 \times 1.1 p_h).$$

$$p_s = 840 \text{#/in}, \quad p_v = 100.15 \text{#/in},$$

$$p_h = 627.8 \text{#/in},$$

$$G\omega_{uA} = 0.75 (627.8 \text{#/in} \times 1.7 \times 1.1) = 881 \text{#/in}$$

$$G\omega_{uB} = 0.75 (840 \times 1.7 + 100.15 \times 1.7 \times 1.1) = 1338 \text{#/in}$$

$$R_A = \frac{(881)(10')}{2} + \frac{1/2(457)10'}{3} = 5167 \text{#}$$

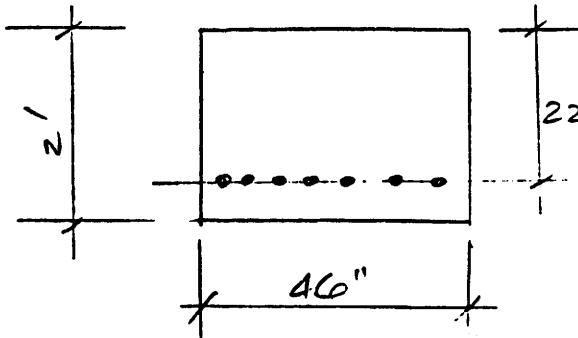
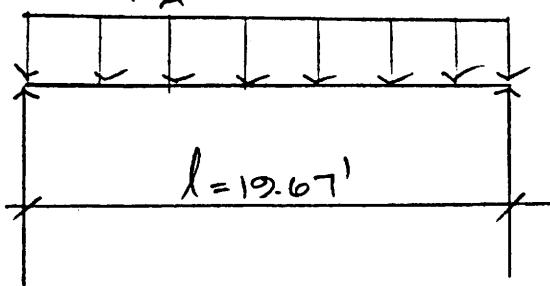
$$R_B = \frac{881(10)}{2} + 2285 \times \frac{2}{3} = 5928 \text{#}$$

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$$R_A = 5167 \text{ #/l}$$



$$M_{ue} = \frac{R_A l^2}{8} = \frac{(5167)(19.67)^2}{8} = 250 \text{ k-in}$$

ASSUME $f'_c = 3 \text{ ksi}$ $f_y = 40 \text{ ksi}$

$$F = \frac{bd^2}{12,000} = \frac{46(22)^2}{12,000} = 1.86$$

$$K_u = \frac{M_u}{F} = \frac{250 \text{ k-in}}{1.86} = 134.4$$

$$\rho = 0.004$$

$$A_s = \rho bd = 0.004(46) \times 22 = 4.04 \text{ in}^2$$

USE 5# 8 & 2-#7
 $A_s = 3.93 \text{ in}^2 + 1.20 = 5.13 \text{ in}^2$

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CHECK LIMITS: ρ .

$$\rho = 0.004$$

$$\rho_b = 0.85^2 \frac{f'_c}{f_y} \frac{87}{87+f_y} = 0.85^2 \frac{3}{40} \frac{87}{127} = 0.0371$$

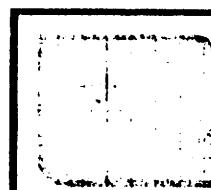
$$\rho_{max} = 0.75 \rho_b = 0.75 (0.0371) = 0.0278 > 0.004 (\text{O.K.})$$

CHECK BEAM SHEAR:

$$R_b = 5928 \text{ #}$$

$$V = \frac{R_b}{\phi bd} = \frac{5928}{0.85(46)(22)} = 6.89 \text{ psi}$$

$$\text{ALLOWABLE SHEAR} = 2\sqrt{f'_c} = 2\sqrt{3000} = 109 \text{ psi} > 6.89 \text{ psi}$$



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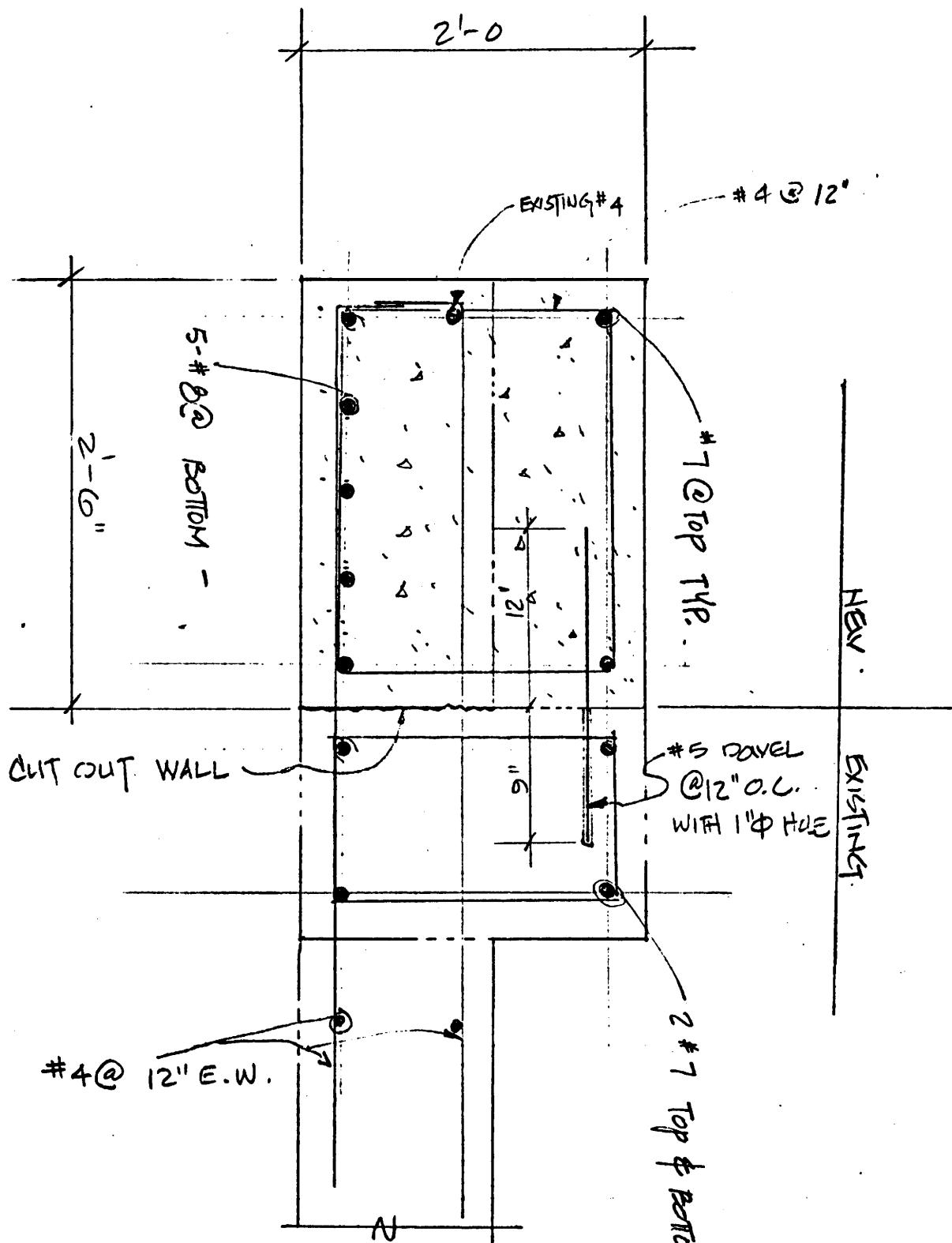
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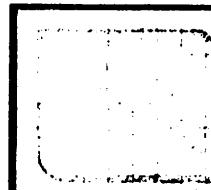
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SECTION



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SUBJECT L.L.N.L.
BLDG. 313

SHEET NO. 30 OF _____
JOB NO. _____

SLAB CAPACITY OVER BASEMENT

GENERAL ASSUMPTIONS

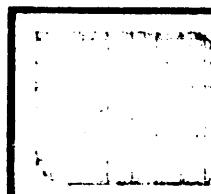
- (1) ASSUME COL HINGE AT TOP, BECAUSE THE COL STIFFNESS IS VERY SMALL WHEN COMPARE WITH SLAB-BEAM STIFFNESS.
- (2) THE DISTRIBUTION OF PANEL MOMENTS ARE BASED ON U.B.C. SECTION 2613(d)
- (3) CONCRETE STRENGTH $f_c' = 3000 \text{ PSI}$
REINF. STEEL $f_y = 40,000 \text{ PSI}$

CHECK C1, C2

USE $1^k/\text{LF}$ LOADINGS

TO FIND C_k

$$M = C_k W l^2$$



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BY A. SETO DATE
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SUBJECT LLNL
BLDG 313

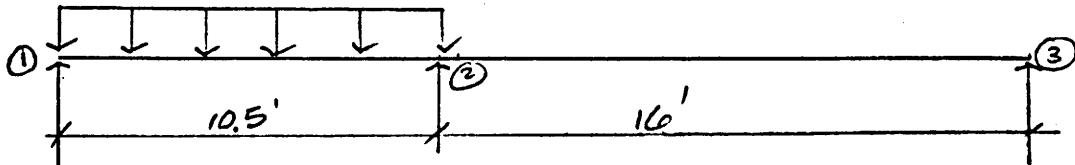
SHEET NO. 31 OF ..
JOB NO.

C1, C2

CASE 1.

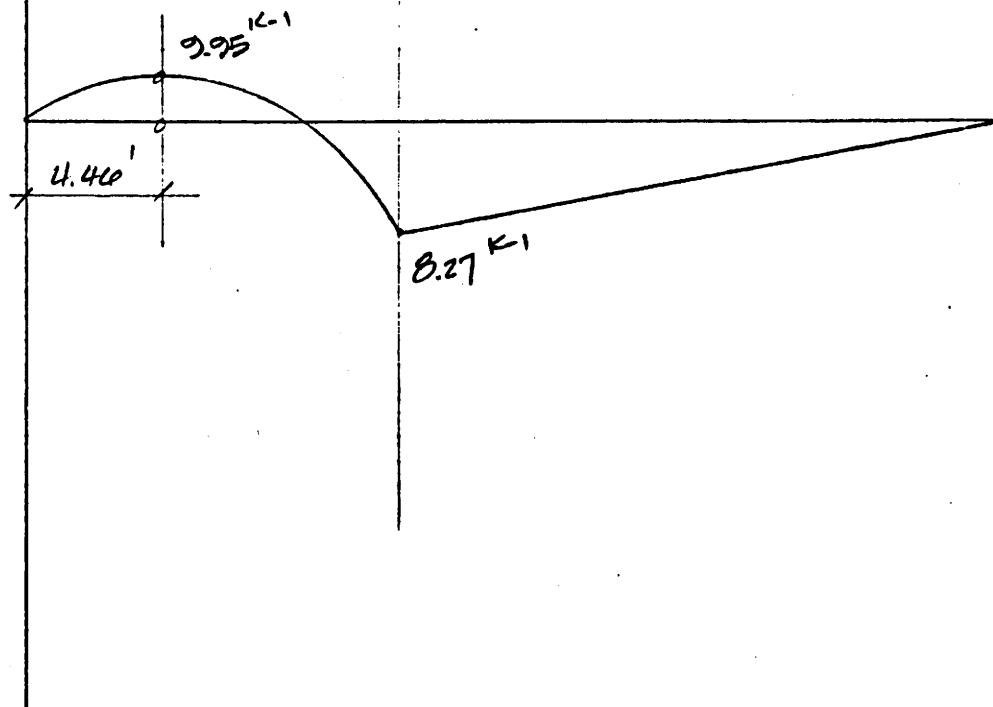
141

$$F.E.M = \frac{w l^2}{8} = \frac{11(10.5)^2}{8} = 13.78$$



O.F.
F.E.M.

$$\begin{array}{r} 24 \text{ O.O} \\ + 13.78 \text{ O} \\ - 5.51 - 8.27 \\ \hline + 0.27 - 8.27 \end{array}$$



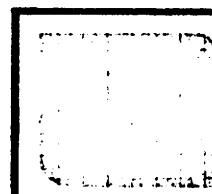
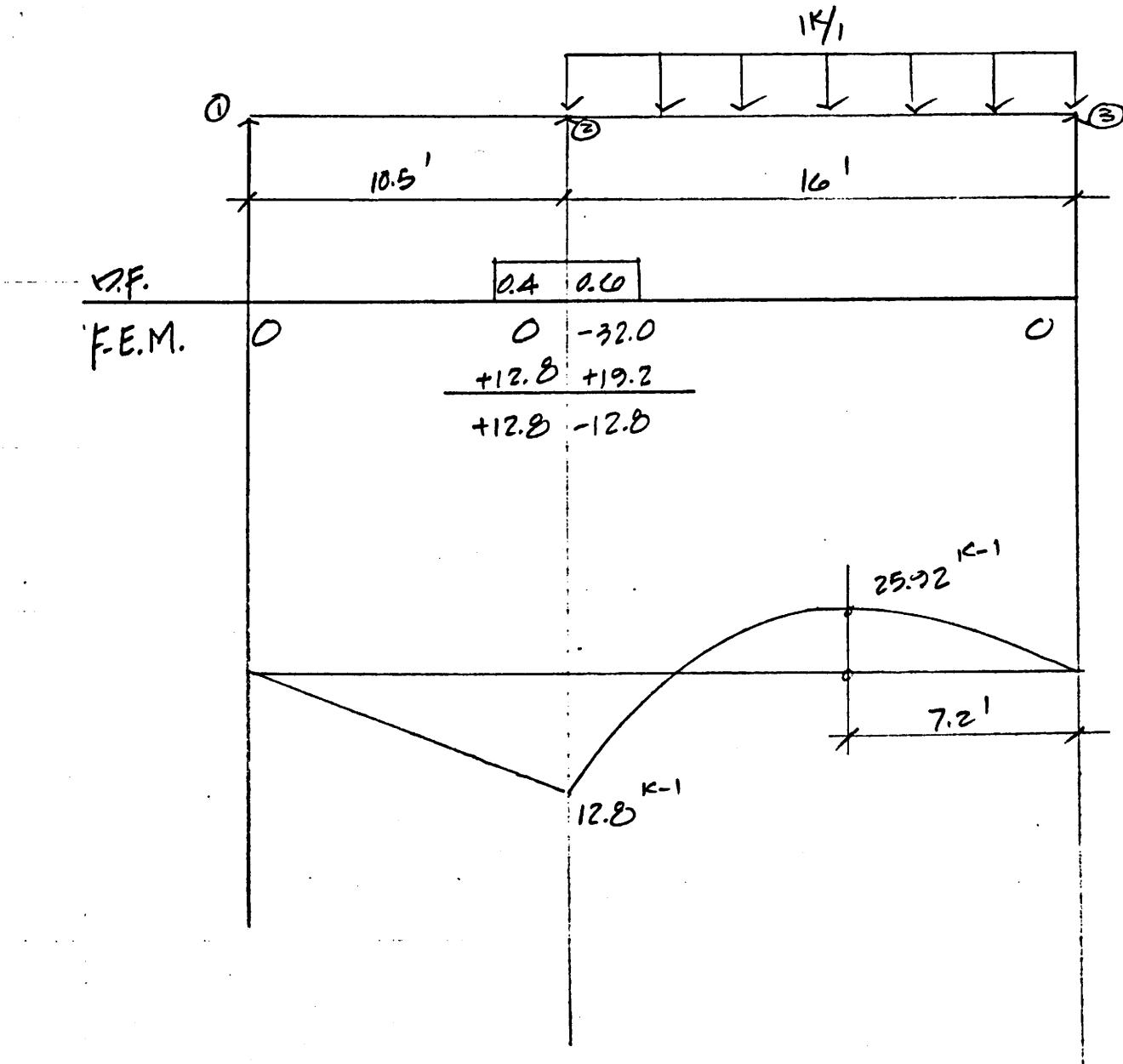
BY A. SETO DATE _____
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SUBJECT LLNC
BLOG 313

SHEET NO. 32 OF _____
JOB NO. _____

C1, C2

CASE 2.



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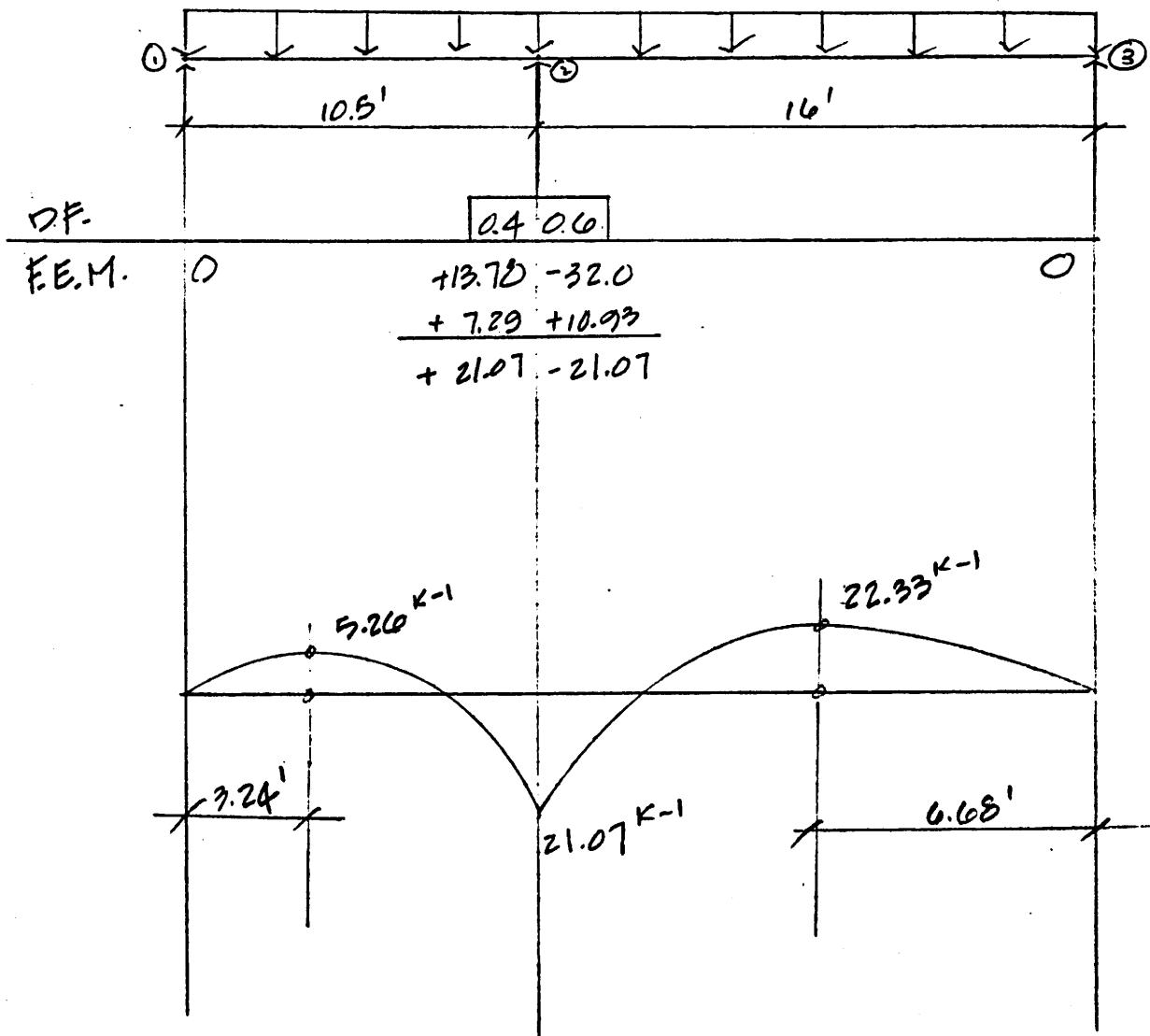
SUBJECT C L N C
BLDG. 313

SHEET NO. 33 OF _____
JOB NO. _____

C1, C2

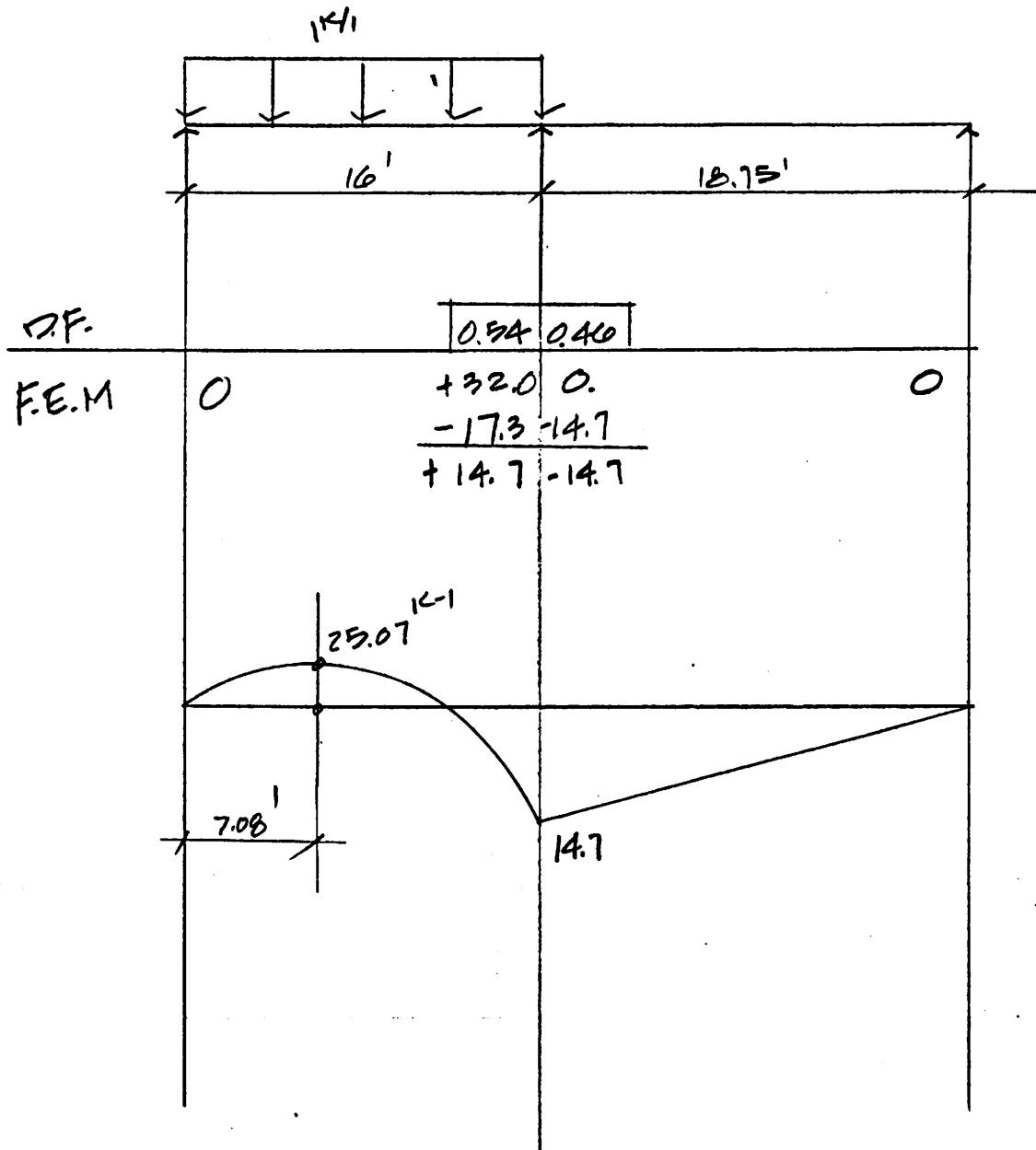
CASE 1 + 2.

1K/1



C5, C7

CASE 1.



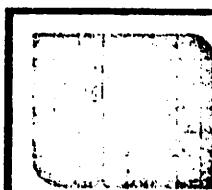
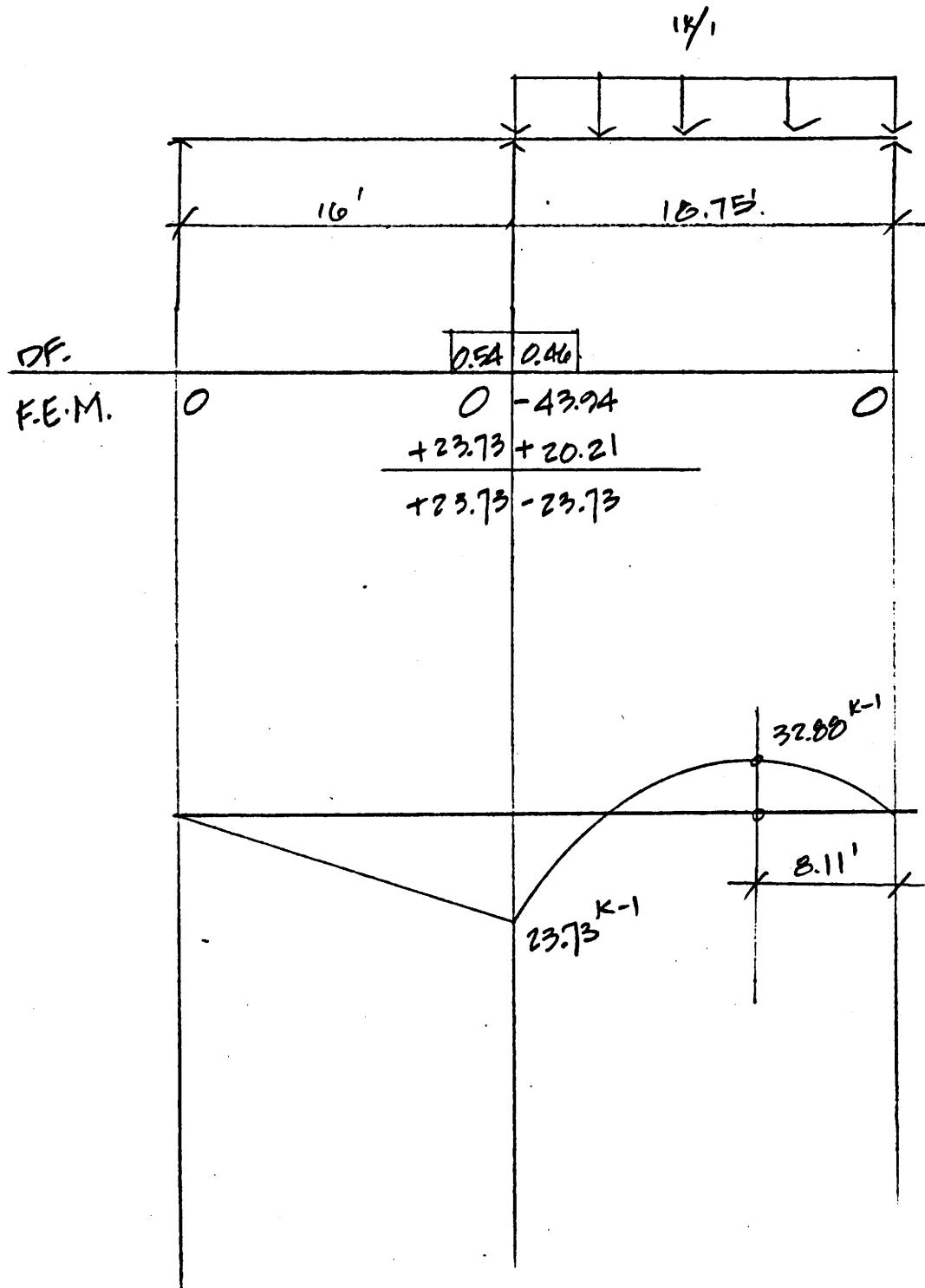
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SUBJECT L LNC
BLDG 312

SHEET NO. 35 OF _____
JOB NO. _____

CS, C7

CASE 2.



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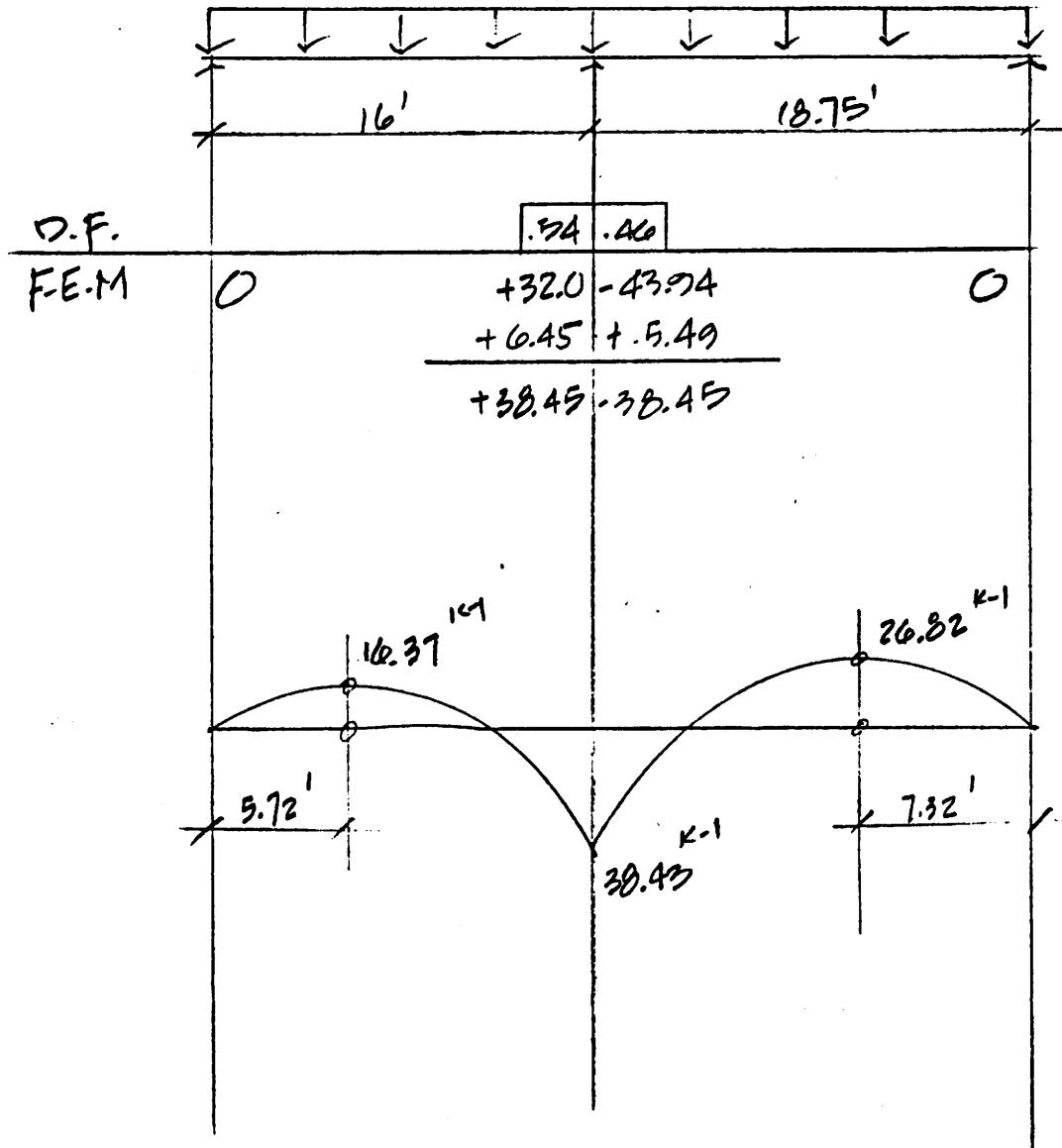
BY A. SETO DATE _____
CHKD. BY _____ DATE _____

SUBJECT L C N C
BLDG 313

SHEET NO. 36 OF _____
JOB NO. _____

C5, C7

CASE 1 + 2.



BY A. SETO DATE
CHKD. BY DATE

SUBJECT LCN
BLDG. 313

SHEET NO. 37 OF
JOB NO.

$$\text{MAX. MOMENT} = 25.92^{\text{k}} \text{ (Case 2)}$$

$$C_R = \frac{25.92}{(16)^2} = 0.1$$

C5, C7

$$\text{MAX MOMENT} = 38.43^{\text{k}}$$

$$C_R = \frac{38.43}{(18.75)^2} = 0.11 \leftarrow \text{GOVERNS}$$

MOMENT CAPACITY

C1, C2

$$A_s = 0.41^{\text{in}^2} (\#5 @ 9^{\text{in}})$$

$$d = 11 - 1.5 = 9.5^{\text{in}} (\text{WHEN } D = 11^{\text{in}})$$

$$M_{u1} = 3.0 A_s d - 2.35 A_s^2 \text{ (ACI HANDBOOK TABLE 7.1.2)}$$

$$= 3.0(0.41 \times 9.5) - 2.35(0.41)^2$$

$$= 11.29^{\text{k}} \text{ (CRITICAL AT COL SUPPORT)}$$

$$M_u @ \text{COL STRIP} = \frac{11.29}{0.75} = 15.05^{\text{k}}$$

$$C_R = 0.1$$

$$W = \frac{15.05}{0.1(16)^2} = 0.59^{\text{k/in}} = 590^{\text{#/ft}}$$

$$DL = \left(\frac{14}{12} \times 150\right) 1.4 = 245^{\text{#/ft}}$$

$$LL = \left(\frac{W - DL}{1.7}\right) = \frac{590 - 245}{1.7} = 202.9^{\text{#/ft}}$$

SAY 200 $\frac{\text{#}}{\text{ft}}$

BY A. SETO DATE.....
CHKD. BY DATE.....

SUBJECT LLNL
BLDG 313

SHEET NO. 30 OF.....
JOB NO.....

C5, C7

MAX MOMENT @ COL. SUPPORT.

$$\#5 @ 9" \quad A_s = 0.41$$

$$d = 14 - 1.5 = 12.5"$$

$$M_u = 3.0 A_s d - 2.35 A_s^2$$

$$= 3.0 \times 0.41 \times 12.5 - 2.35 (0.41)^2$$

$$= 14.98 \text{ k} \quad \frac{14.98}{0.75} = 19.97 \text{ k}$$

$$C_k = 0.15$$

$$w = \frac{M}{C_k L^2} = \frac{19.97}{0.11 \times (18.75)^2} = 0.52 \text{ psf}$$

$$LL = \frac{520 - 245}{1.7} = 161.75 \text{ psf}$$

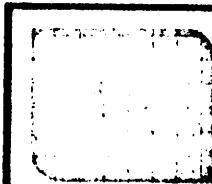
SAY 160 #/sf

$$WHE \quad d = 9.5$$

$$A_s = 0.59 \quad \#6 @ 9"$$

$$M_u = 3.0 \times 0.59 \times 9.5 - 2.35 (0.59)^2$$
$$= 16.11 \text{ k} > 14.98$$

$\therefore D = 14"$ CONTROL



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SHEET NO. 29 OF _____
JOB NO. _____

CHECK SHEAR

$$DL = 175 \times 1.4 = 245$$

$$LL = 160 \times 1.7 = 272$$

$$\underline{517 \text{ PSF}}$$

CHECK TWO-WAY SLAB

$$\text{COL SIZE } 14'' \times 14'' \quad d = 9.5''$$

$$b_0 = (14 + 9.5) 4 = 94''$$

$$V_U = 0.517 (13.33 \times 17.625 - \overline{1.167}^2) \\ = 120.76^k$$

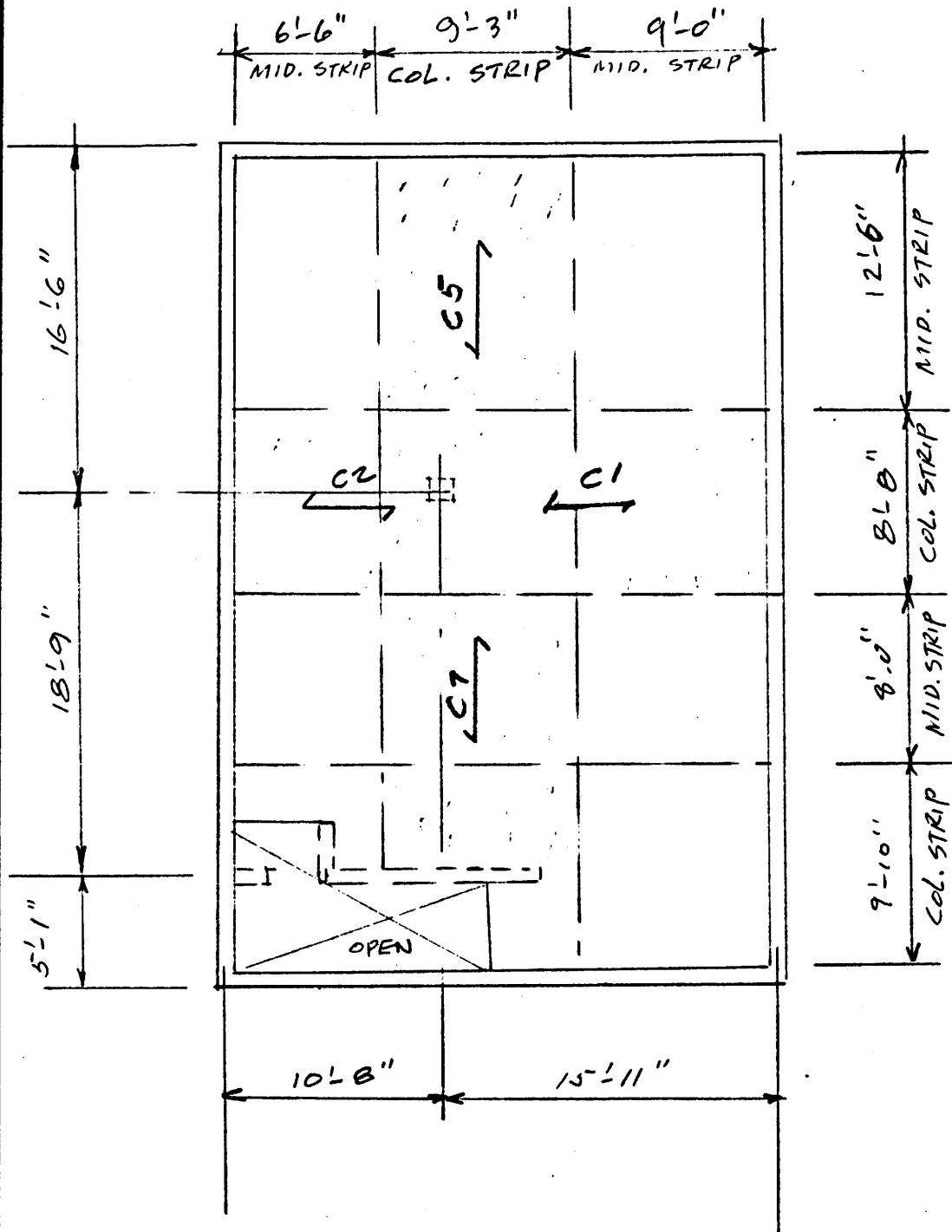
$$\phi V_c = 0.85 (\sqrt{f'_c} b d) \\ = 0.85 (4 \sqrt{3000} \times 94 \times 9.5) / 1000 \\ = 166.3^k > V_U = 120.76^k \text{ O.K.}$$

CHECK BEAM SHEAR

$$V_U = 0.517 \times 8' = 4.14^k$$

$$\phi V_c = 0.85 \times 2 \sqrt{f'_c} b d \\ = 0.85 \times 2 \sqrt{3000} \times 12 \times 9.5 / 1000 \\ = 10.61^k > V_U = 4.14^k \text{ O.K.}$$

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PLAN = SLAB ABOVE BASEMENT

BY DATE
CHKD. BY DATE

SUBJECT LLNC
..... MISSILE

SHEET NO. 40 OF
JOB NO. 1354

MISSILE IMPACT

ASSUMPTIONS =

1. MISSILE 2" x 4" x 12^{1/2}o TIMBER, WT 20 lb, TRAVELS @ 70 M.pH.

PENETRATION @ REINFORCING CONC.

REF.: MODIFIED PETEY FORMULA.

$$x = 12 K_p A_p \log_{10} \left[1 + \frac{V_s}{215,000} \right]$$

$$x_1 = \left[1 + e^{-4(\frac{t}{2} - 2)} \right] x, \quad (t > 2x)$$

WHERE x = DEPTH OF MISSILE PENETRATION INTO CONC. ELEMENT OF INFINITE THICKNESS. (INCHES)

K_p = PENETRATION COEFFICIENT FOR REINFORCED CONC. (FIG. 2-1)

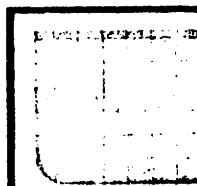
$A_p = \frac{W}{A} = \frac{\text{MISSILE WT}}{\text{PROJECTED FRONTAL AREA OF MISSILE}}$ (PSF)

V_s = STRIKING VELOCITY OF MISSILE CFT/SEC.)

x_1 = DEPTH OF MISSILE PENETRATION INTO CONC. ELEMENT OF FINITE THICKNESS (INCHES)

e = BASE OF NAPIERIAN LOGARITHMS

t = THICKNESS OF CONC. ELEMENT. (INCHES)



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SHEET NO. 41 OF _____
JOB NO. _____

$$K_p = 0.0035$$

$$A_p = \frac{20 \text{ lb}}{0.167 \times 0.33} = 363.64 \text{ lb/ft}^2$$

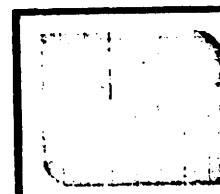
$$V_s = \frac{70 \text{ mph} \times 5280}{60 \times 60} = 102.67 \text{ ft/sec}$$

$$\therefore x = 12 \times 0.0035 \times 363.64 \log_{10} \left[1 + \frac{102.67^2}{215000} \right]$$

$$x = 0.32''$$

$$x_1 = \left[1 + e^{-4(\frac{8}{2} - 2)} \right] 0.32$$

$$x_1 = 0.32'' < 8'' \text{ O.K.}$$



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SHEET NO. 42 OF _____
JOB NO. _____

MISSILE (CONT.)

PERFORATION @ CONC. WALL

REF. BALLISTIC RESEARCH LABORATORY, BRL FORMULA

$$T = \frac{427}{\sqrt{f'_c}} \frac{kl}{D^{1.8}} \left(\frac{V_s}{1000} \right)^{1.33}$$

$$t_p = 1.25T \leq T + 10 \text{ (INCHES)}$$

WHERE T = THICKNESS OF CONC. ELEMENT TO BE JUST
PERFORATED (INCHES)

k = WT. OF MISSILES (lb)

D = DIAMETER OF MISSILES (INCHES)

V_s = STRIKING VELOCITY OF MISSILE (ft/sec.)

f'_c = COMP. STRENGTH OF CONC. (3000 psi)

t_p = CONC. ELEMENT REQ'D TO PREVENT
PERFORATION.

$$D = \text{EQUIV. DIAM.} = \sqrt{\frac{4A}{\pi}} = \sqrt{\frac{4 \times 8}{\pi}} = 3.19 \text{ INCH}$$

$$\therefore T = \frac{427}{\sqrt{3000}} \times \frac{20}{3.19^{1.8}} \left(\frac{102.67}{1000} \right)^{1.33}$$

$$T = 0.935"$$

$$t_p = 1.25 \times 0.935 = 1.17" < 8" \text{ O.K.}$$

BY P.L.W. DATE 7/91
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SHEET NO. 43 OF 43
JOB NO. _____

MISSILE (CONT.)

SPALLING @ CONC. WALL

$$T_s = 2T$$

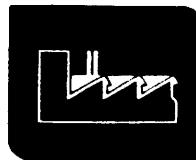
$$t_s = 1.25 T_s \leq T_s + 10 \text{ (INCH)}$$

WHERE T_s = CONC. ELEMENT THICKNESS THAT WILL JUST
START SPALLING (INCH)

t_s = CONC. ELEMENT THICKNESS REQ'D TO
PREVENT SPALLING.

$$T_s = 0.935 \times 2 = 1.87 "$$

$$t_s = 1.25 \times 1.87 = 2.34 " < 3" \text{ O.K.}$$



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APPENDIX B

Sketches, Specifications and Construction Drawings

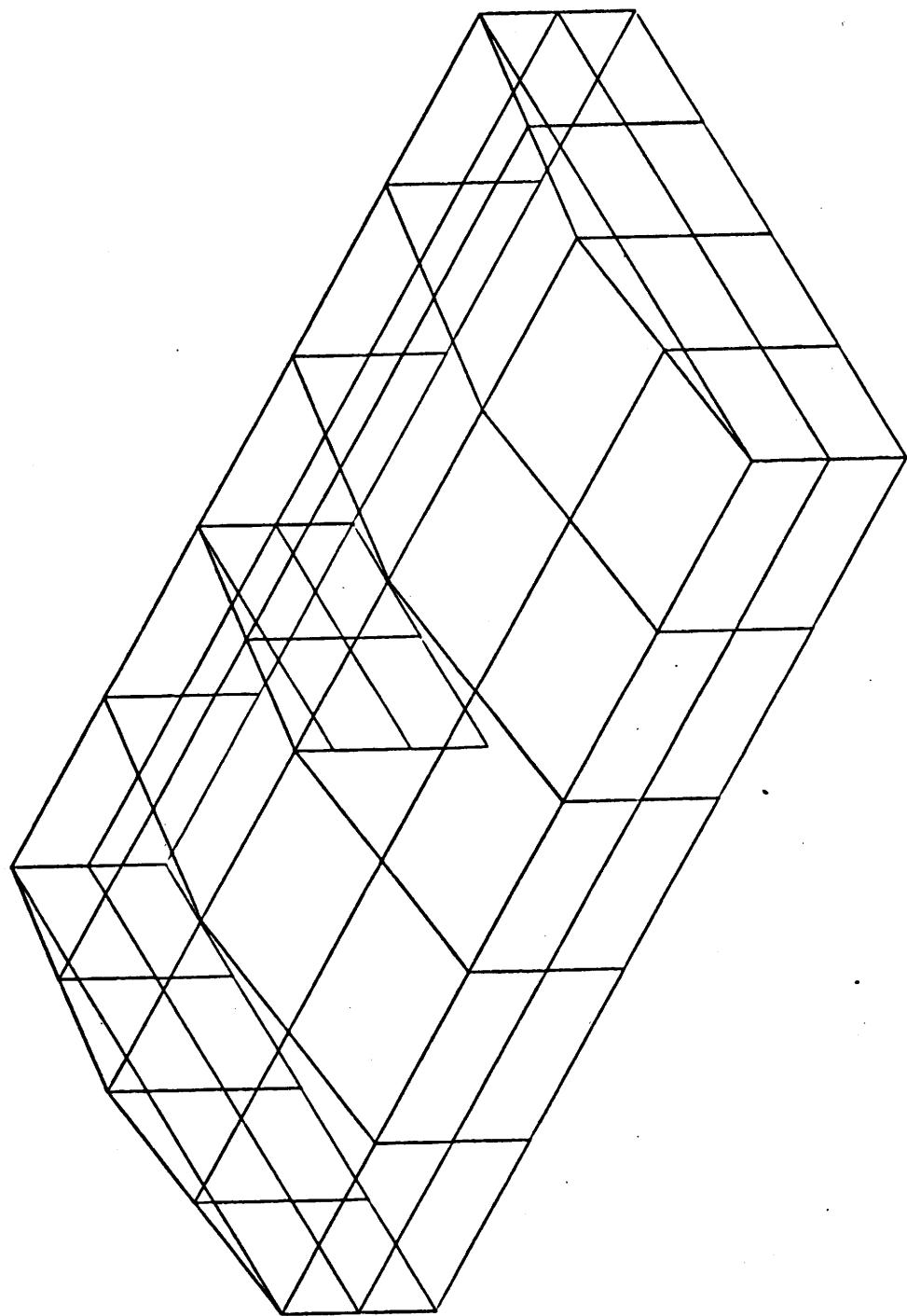
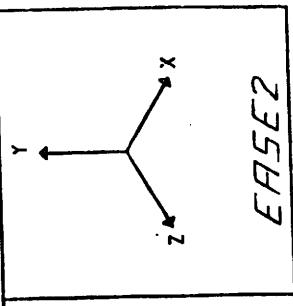


FIGURE I

ONE STORY BUILDING

MODEL SCALE = .7290E-02



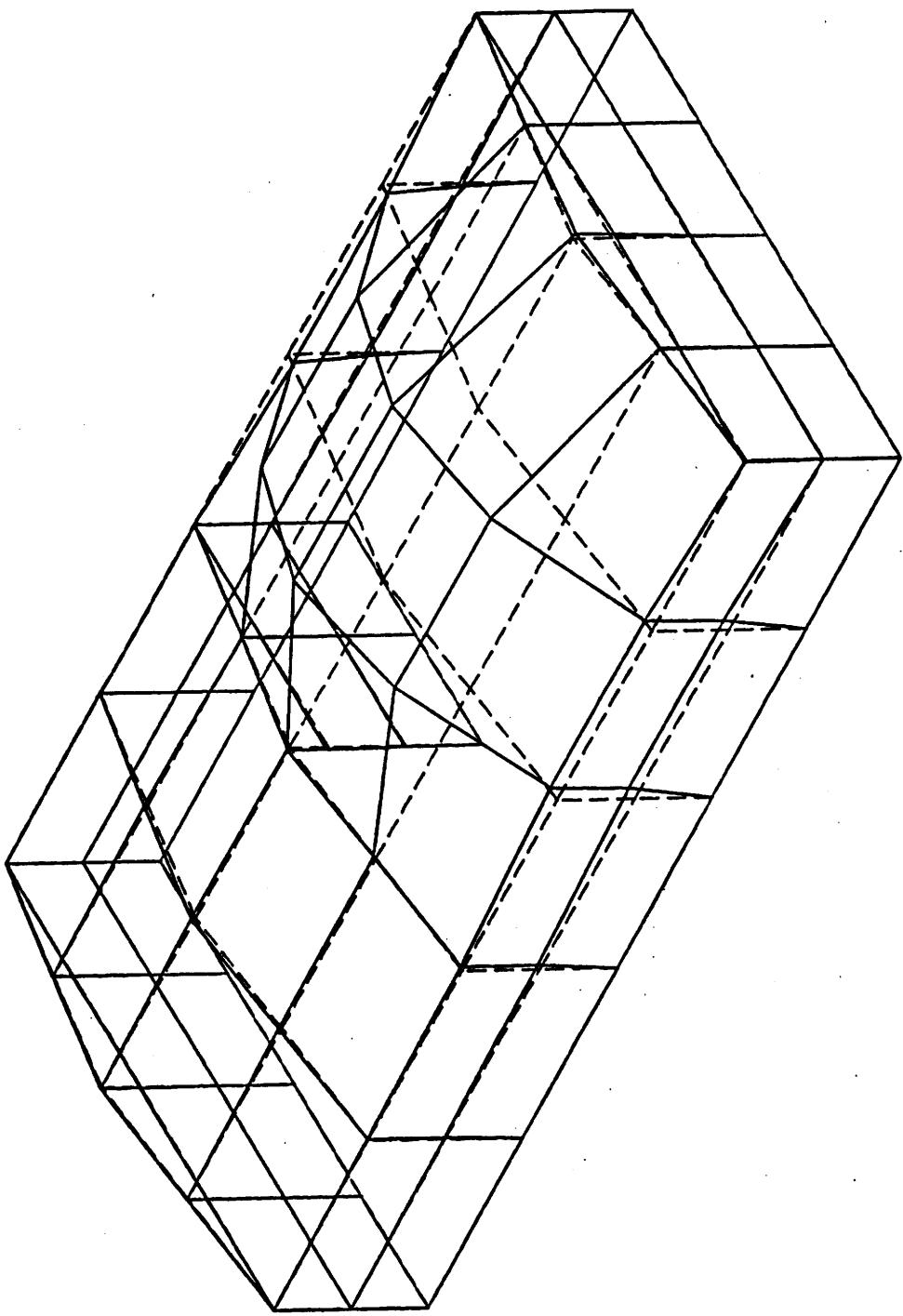
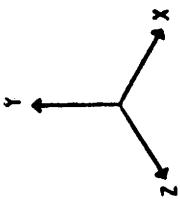


FIGURE 2

MODE SHAPE FOR SINGLE STORY BUILDING

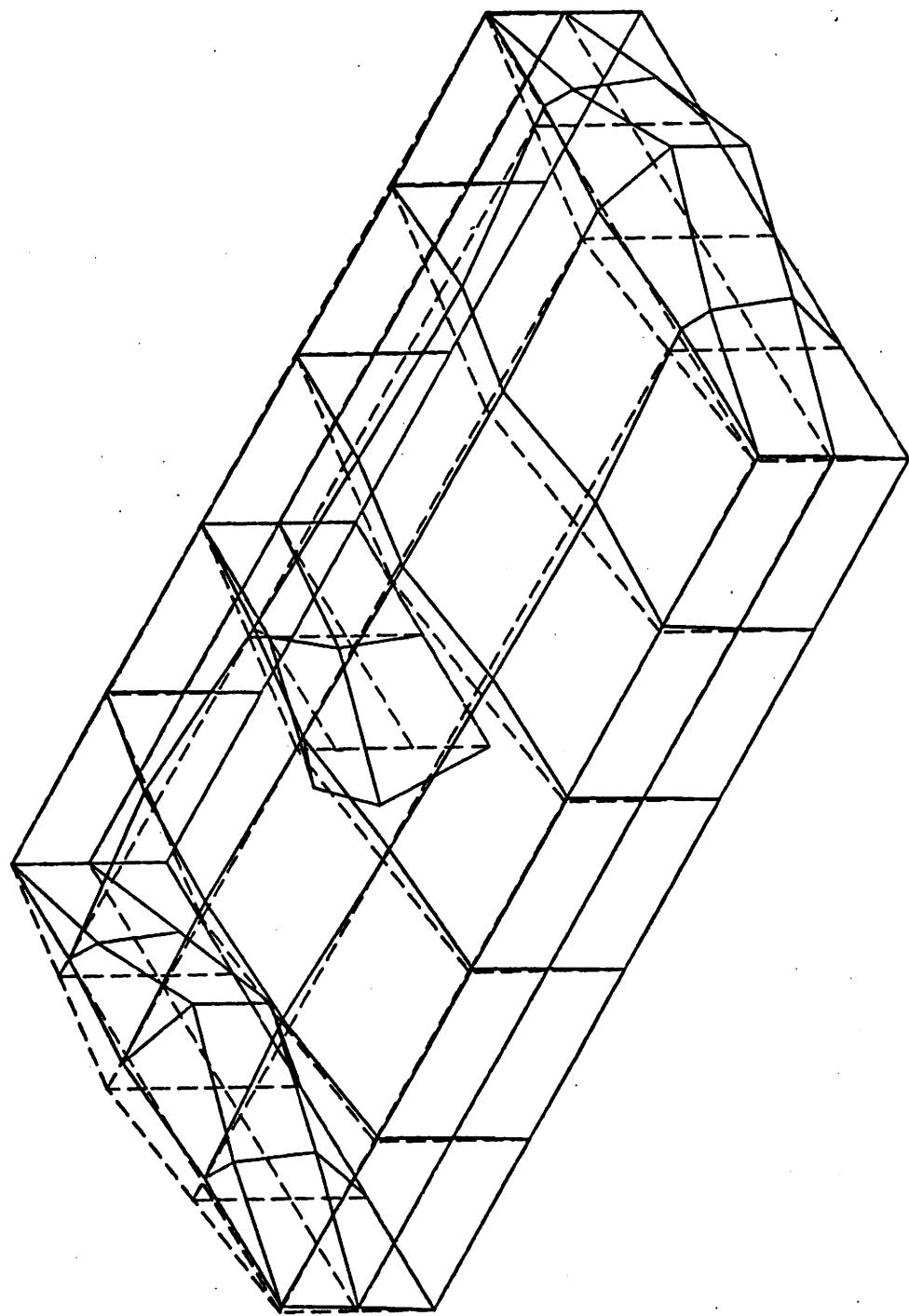
MODEL SCALE = .7290E-02

MODE SHAPE 1 (12.0 HZ)



ERASE2

FIGURE 3



MODE SCALE = .7290E-02 MODE SHAPE 13 (24.7 Hz)

ER5E2

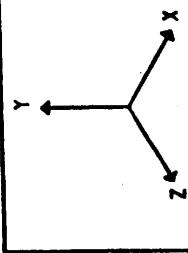
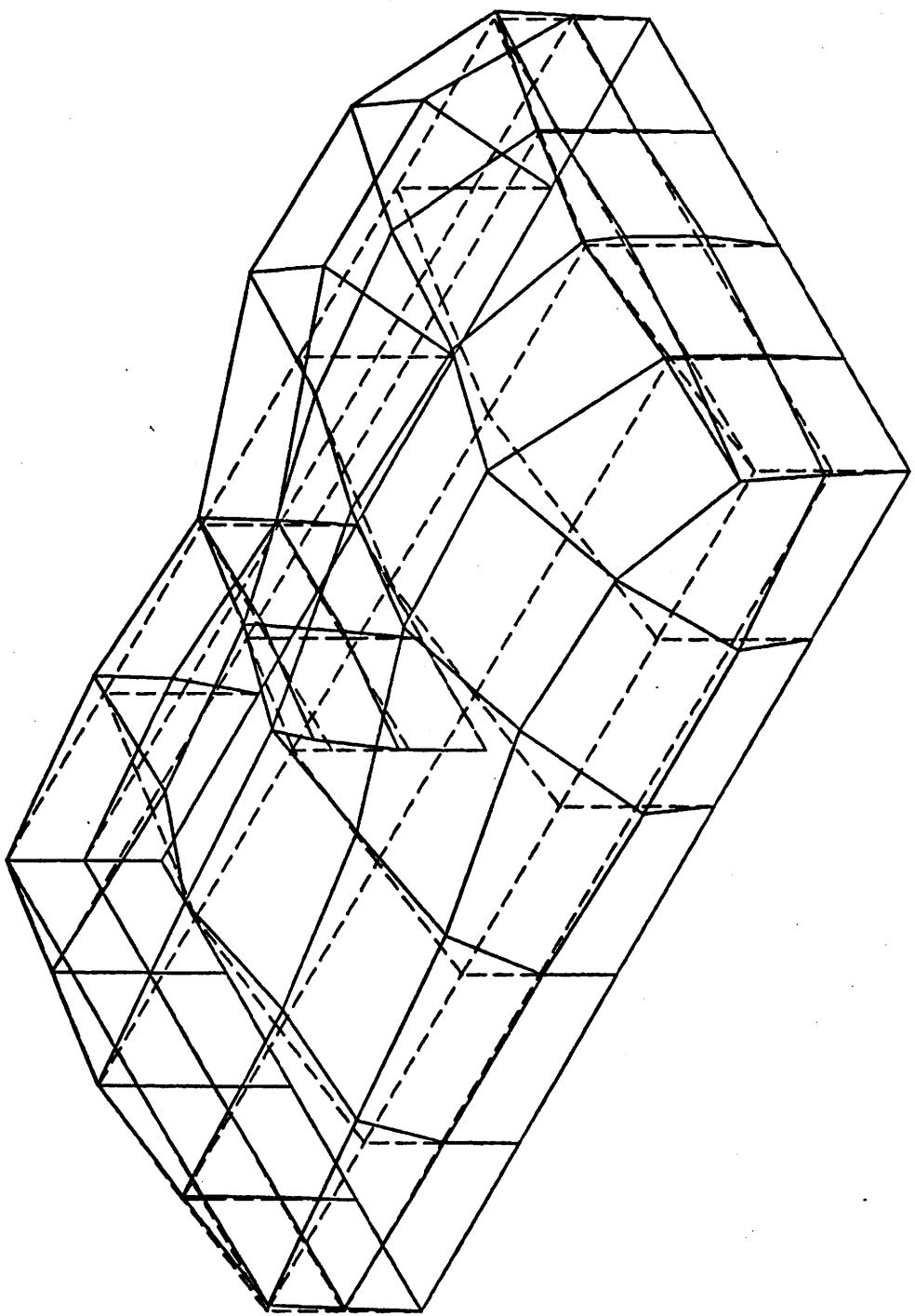


FIGURE 4



MODE SHAPE FOR SINGLE STORY BUILDING

MODEL SCALE = .7290E-02 MODE SHAPE 15 (29.8 Hz)

ERASE2

L.I.L. DESIGN BASIS EARTHQUAKE
HORIZONTAL AND VERTICAL GROUND RESPONSE SPECTRA
PEAK GROUND ACCELERATION NORMALIZED TO 1g.

$$ZPA Q_V = \frac{2}{3} \times ZPA Q_H$$

ZPA: ZERO PERIOD ACCELERATION

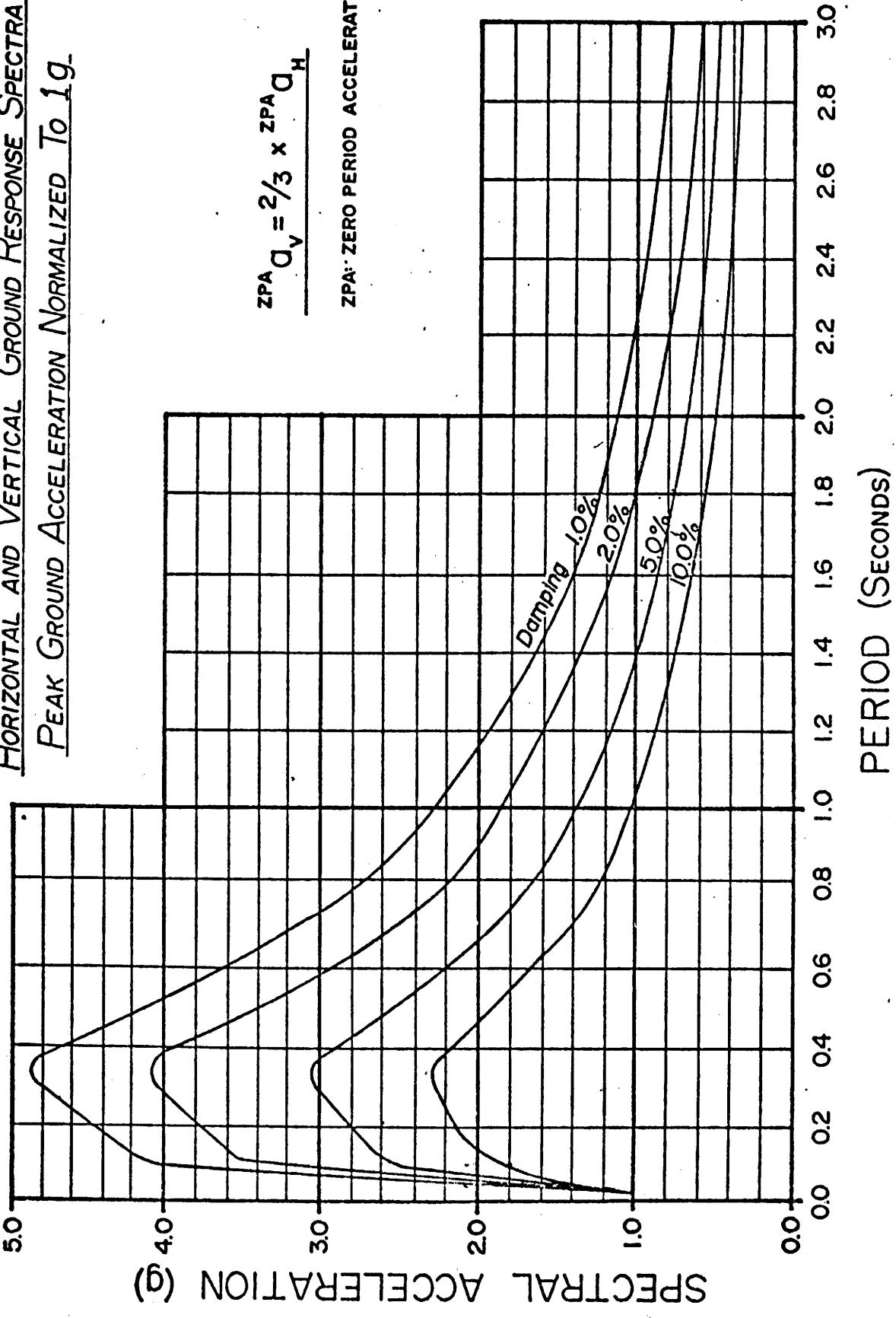


FIGURE 5

GENERAL NOTES

GENERAL

1. The following notes shall apply to all drawings. All work shall be in accordance with requirements of "UNIFORM BUILDING CODE", 1979 Edition, "AISC MANUAL" Seventh Edition and "ACI STANDARD 318-77".

CONCRETE AND REINFORCEMENT

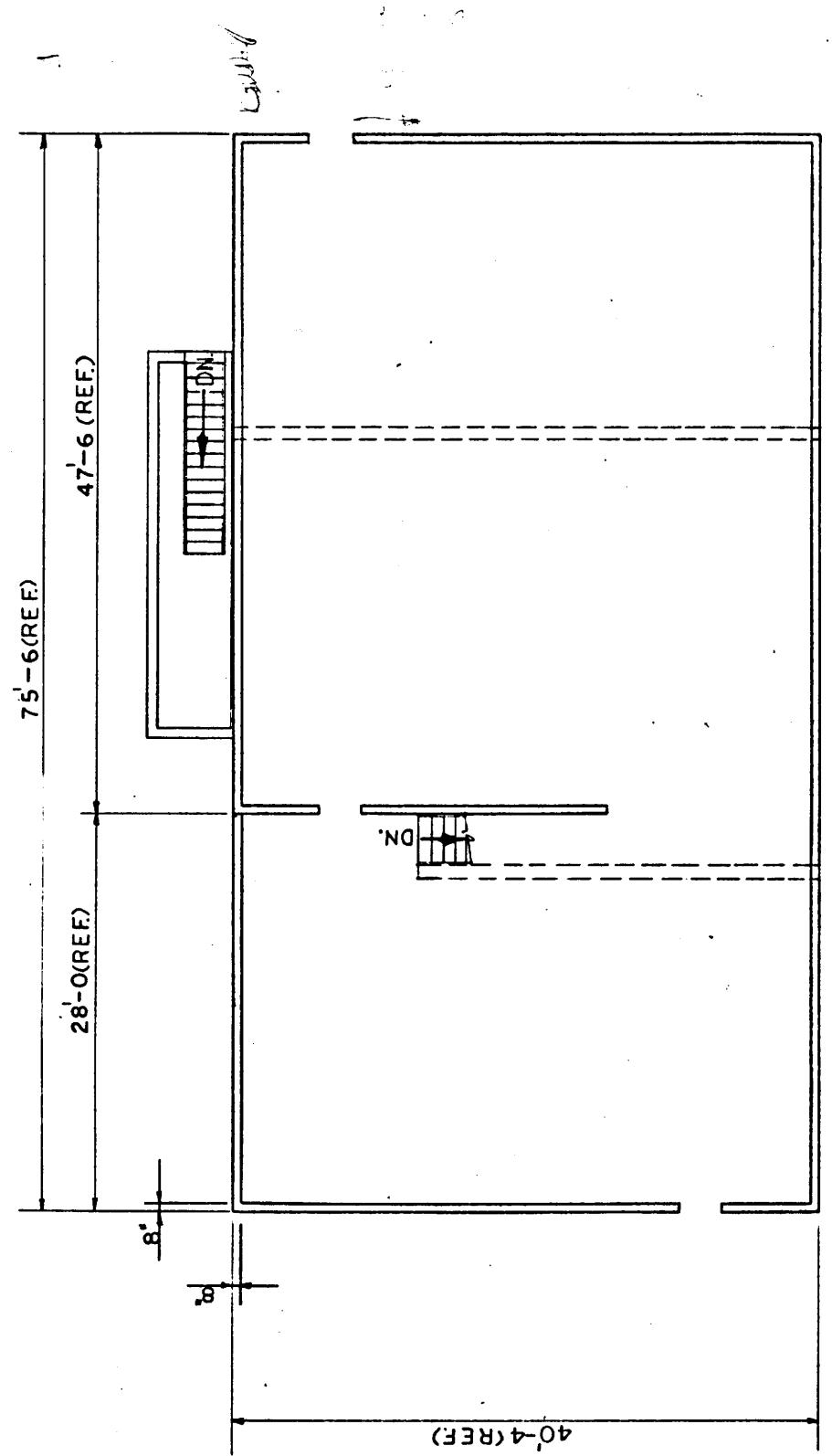
1. Concrete shall have a minimum compressive strength of 3,000 psi at 28th day.
2. Reinforcing steel shall be new, free of scale and rust, per ASTM A-615 Grade 40.
3. Unless otherwise noted, reinforcing steel shall have a minimum concrete cover as per ACI 318-77 Chapter 7.
4. Splices in reinforcement unless otherwise noted 40 dia. of bar.
5. All anchor bolts, shall be "Red Head" Phillip's Anchors or equal. Bolt sizes are shown on the drawings.
6. All exposed edges shall have 3/4" chamfer.

STRUCTURAL STEEL

1. Structural steel, plate and bars shall be ASTM A-36.
2. Connection bolts shall be 3/4" Dia. ASTM A 325 with threads excluded from the shear plane unless otherwise noted.
3. For bolt connection, a minimum of two bolts is required.
4. All welds shall be made with #70 XX electrodes.
5. Details shown are typical. Where specific details are not shown, provide typical connections as specified.
6. All gusset plate shall be 3/8" thick unless noted otherwise.
7. All steel shall be painted one shop coat with red lead FS TT-T-86A Type II, or zinc chromate primer. Do not paint the edges and surface areas where they will be field welded.

△								
△								
△	7-17-81	ISSUED FOR CONSTRUCTION.		PP	SP			
REV.	DATE	REVISIONS	BY	CHK.	DEPT. APPR.	PROJ. ENGR.	CLIENT APPR.	
FREDERIKSEN ENGINEERING CONSULTING ENGINEERS ARCHITECTS				L.L.N.L. DISASTER CENTER BUILDING 313 SEISMIC REINFORCEMENT SPECIFICATIONS				
				DATE 7-17-81 DRAWN BY SCALE N/A CHK. DRAWING NO. REV.				
				1354-A0 0				

FIRST FLOOR PLAN

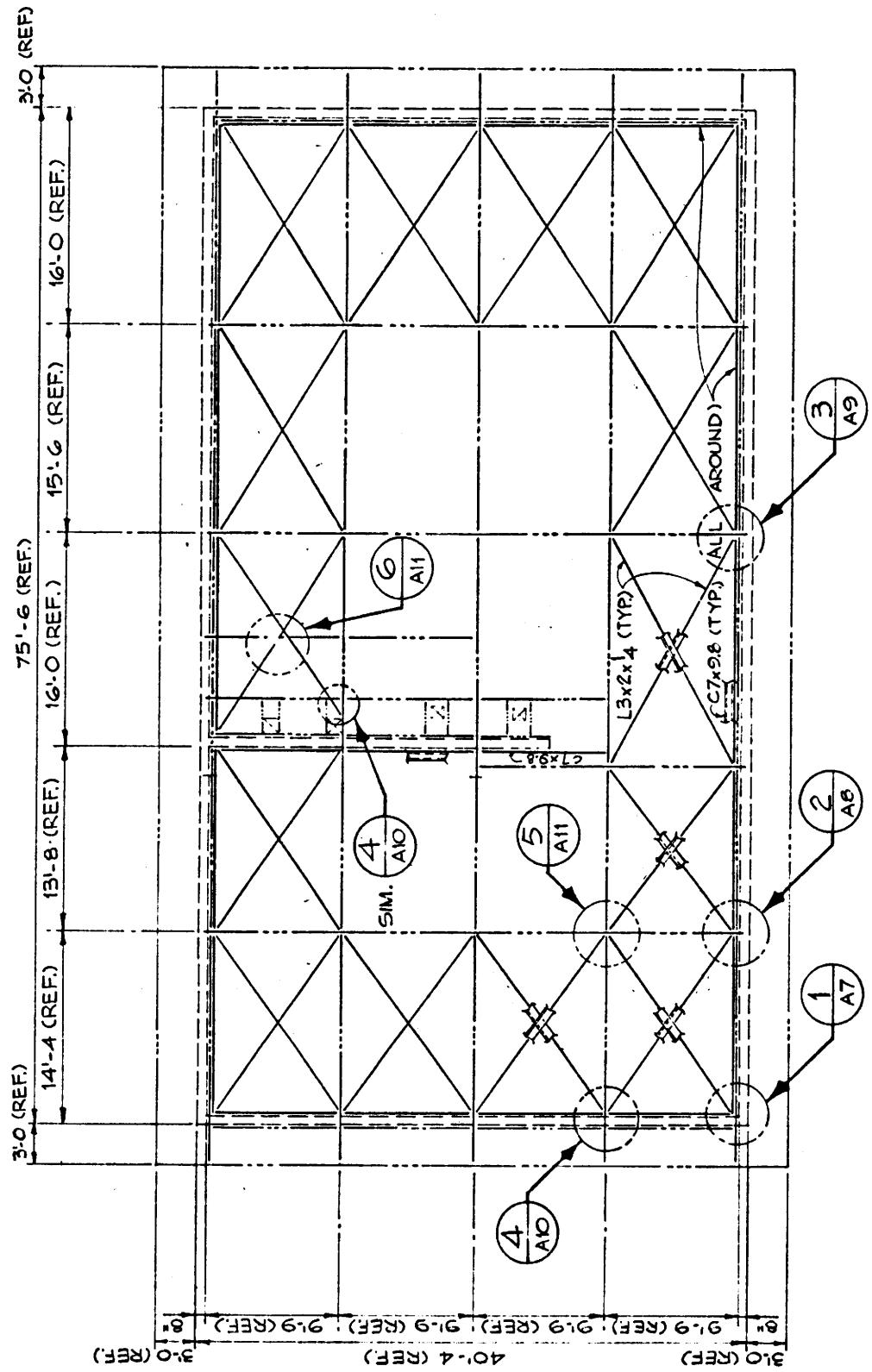


DWG. REDUCED		
DRAWN 7-17-81	DRAWN BY CHAN	SCALE 1:1-0
L.L.N.L. DISASTER CENTER		REV.
BUILDING 313		
SEISMIC REINFORCEMENT		
FIRST FLOOR PLAN		
1354-A1	O	
REVISIONS		
DATE		

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ROOF FRAMING PLAN

DWG. REDUCED		DATE 7-16-81		DRAWN BY DOWNEY		C.H.W.		P.M.J.		SCALE 1/8"=1'-0"	
				L.L.N.L. DISASTER CENTER						DRAWING NO.	
				BUILDING 313						1354-A2	
				SEISMIC REINFORCEMENT						ROOF FRAMING PLAN	
				OAKLAND LONG BEACH							
				CONSULTING ENGINEERS ARCHITECTS							
				FREDERIKSEN ENGINEERING							
				SUED FCR CONSTRUCTION.		BP		E			
REV.	DATE	PROJ. NO.	NAME	CHK.	DEPT.	PROJ. NO.	NAME	CHK.	DEPT.	PROJ. NO.	NAME
			PERMISSIONS								



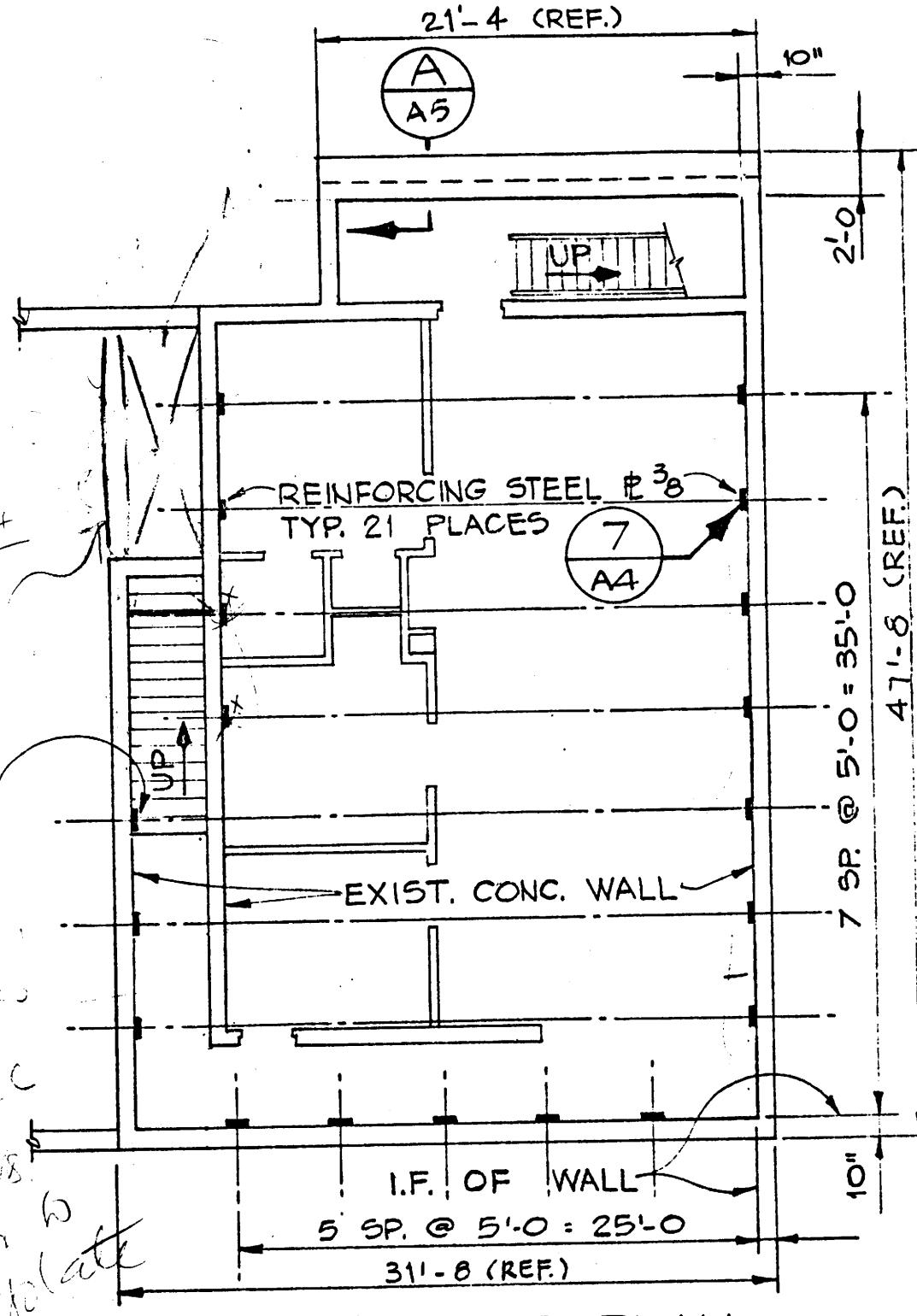
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REINFORCING
STEEL #³/₈
FIELD FIT
AS REQ'D.

BURGUNDY
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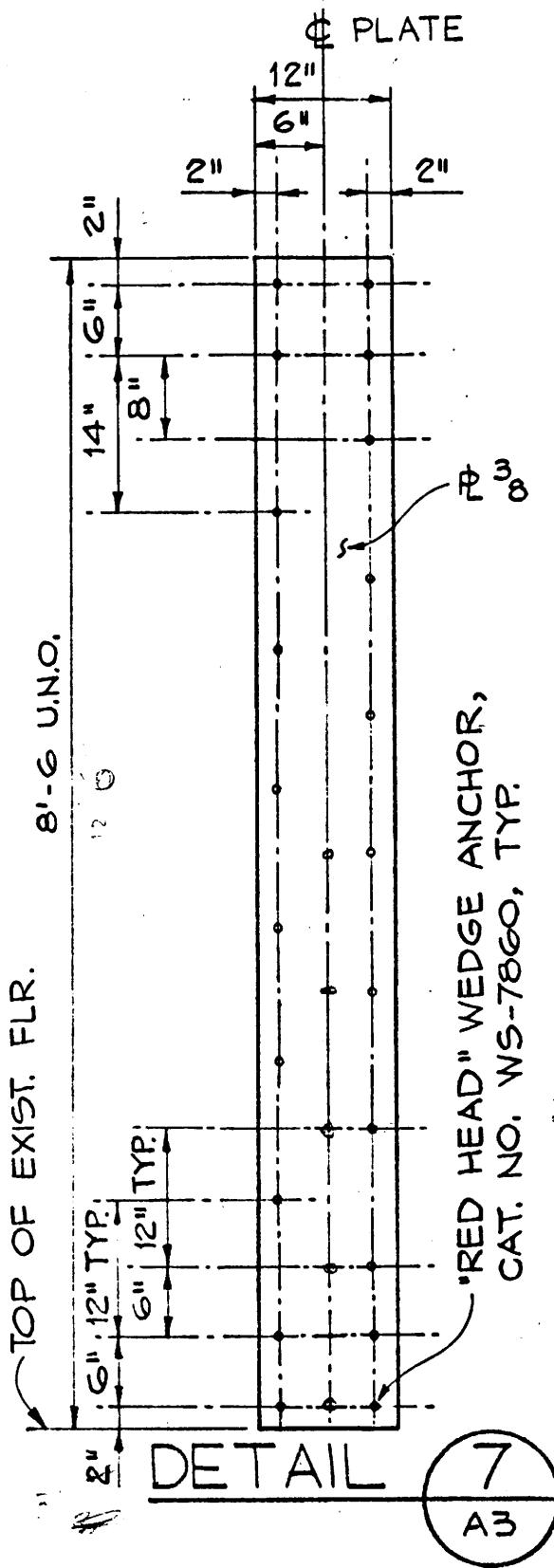
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BASEMENT PLAN

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REV.	DATE	REVISIONS	BY	CHK.	DEPT. APPR.	PROJ. ENGR.
						CLIENT APPR.
FREDERIKSEN ENGINEERING			L.L.N.L. DISASTER CENTER	DATE 7-14-01	DRAWN BY	POWERCO
CONSULTING ENGINEERS ARCHITECTS			BUILDING 313	SCALE 1' ⁸ "=1'-0"	CHK.	PUN
OAKLAND LONG BEACH			SEISMIC REINFORCEMENT	DRAWING NO.		
			BASEMENT PLAN	1354-A3	0	REV.

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"RED HEAD" WEDGE ANCHOR,
CAT. NO. WS-7860, TYP.

BLDG NO. 333 UNIT NO. 2102
11-0-13 2/20/01

REV.	DATE	REVISIONS	BY	CHK.	DEPT.	PROJ.	CLIENT
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SCALE 3/4" = 1:0 CHK. FUN

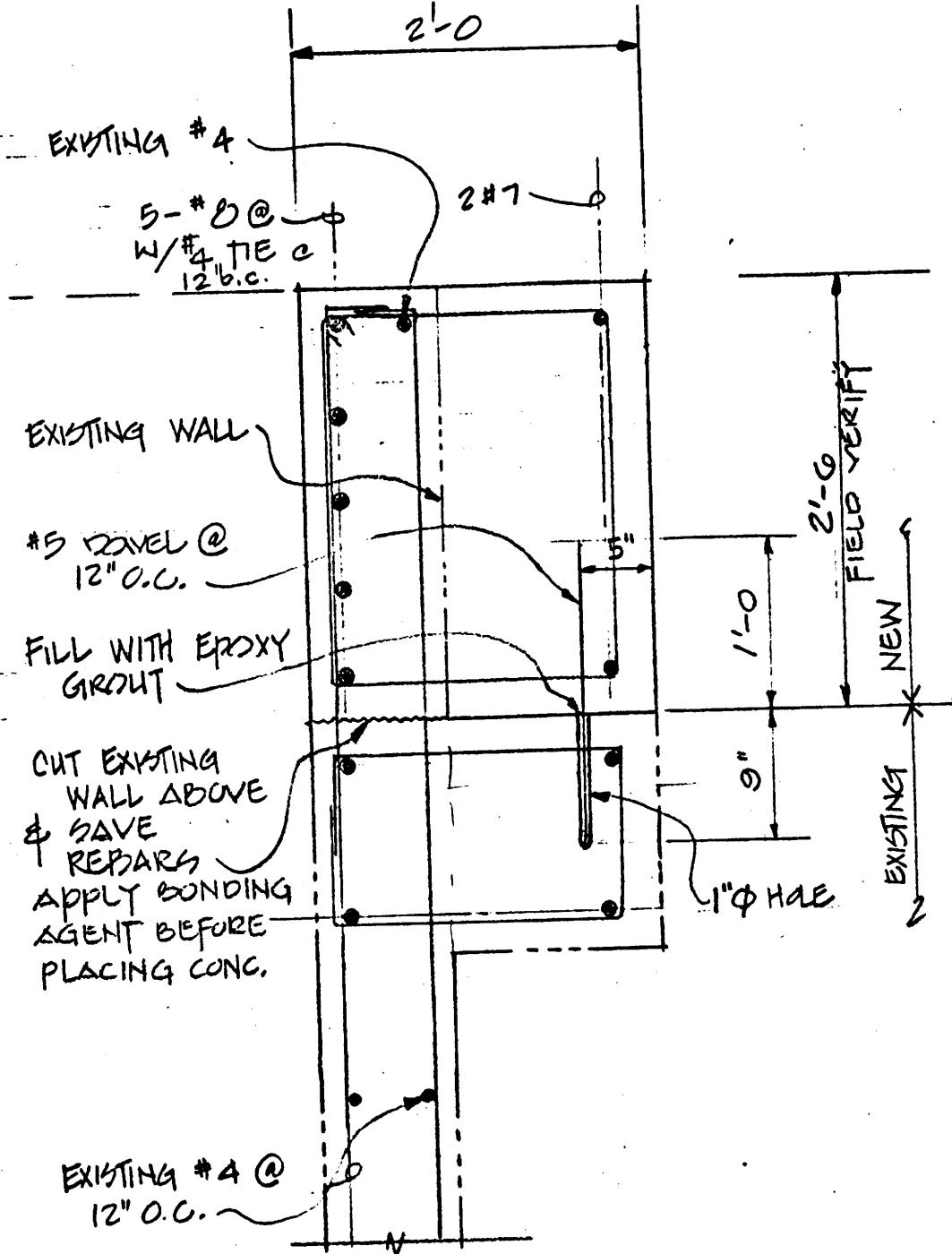
DRAWING NO. REV.

FREDERIKSEN ENGINEERING
CONSULTING ENGINEERS ARCHITECTS OAKLAND LONG BEACH

L.L.N.L. DISASTER CENTER
BUILDING 313
SEISMIC REINFORCEMENT
REINF. STEEL PLATE DET.

1354-A4 0

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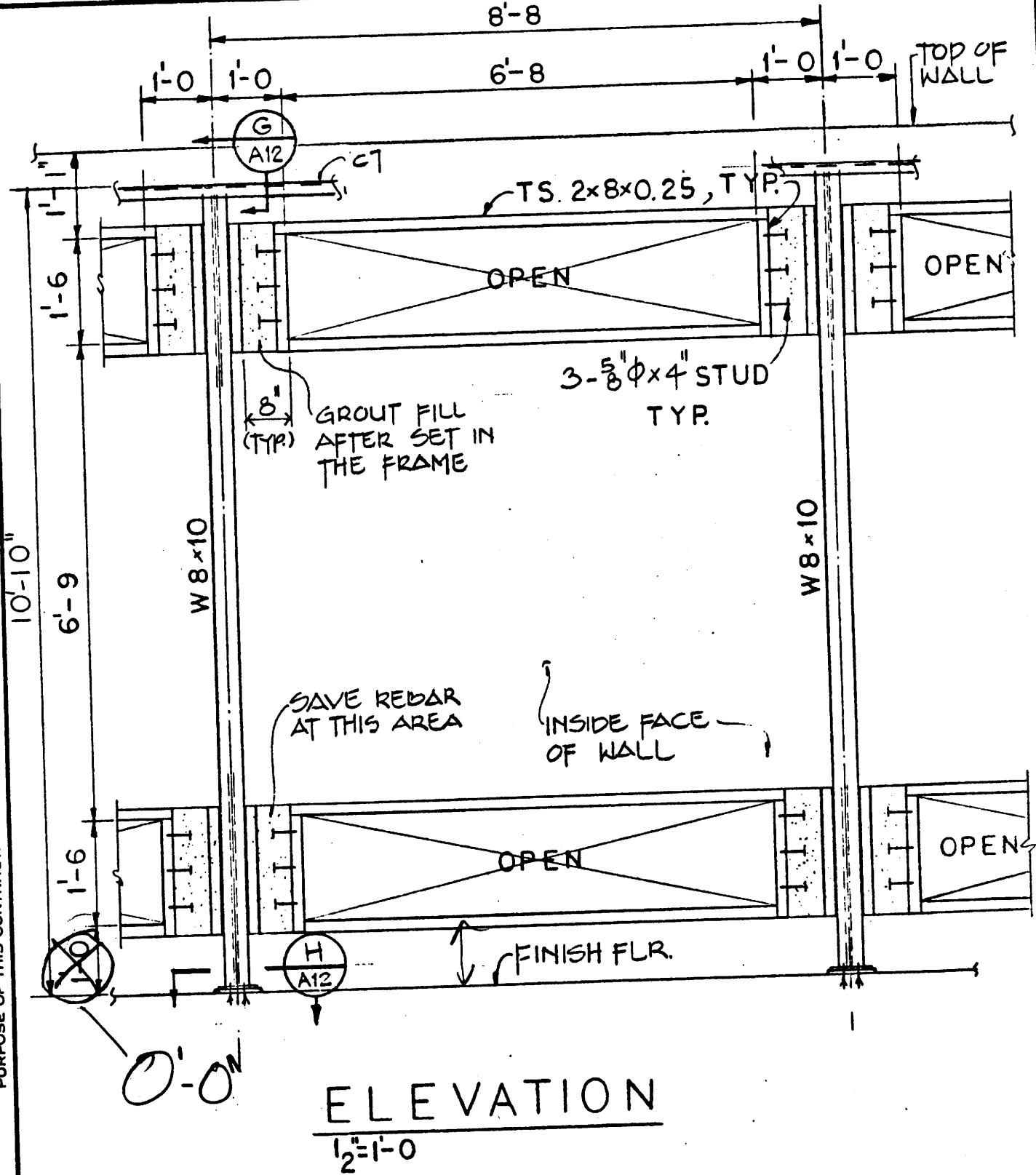


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1" = 1'-0"

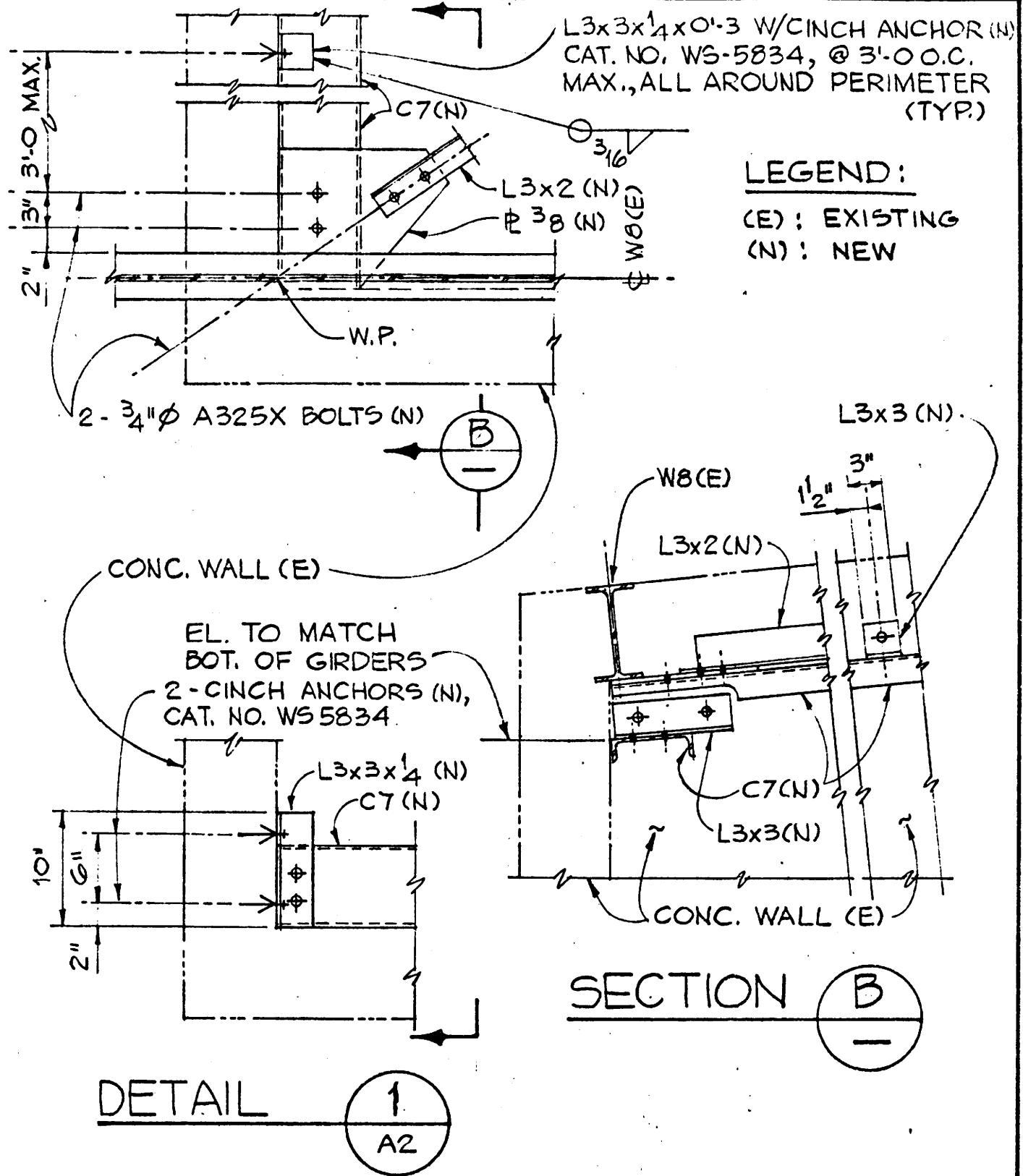
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FREDERIKSEN ENGINEERING			L.L.N.L. DISASTER CENTER BUILDING 313 SEISMIC REINFORCEMENT GRADE BEAM DETAIL					
CONSULTING ENGINEERS ARCHITECTS			OAKLAND LONG BEACH					
DRAWING NO.								
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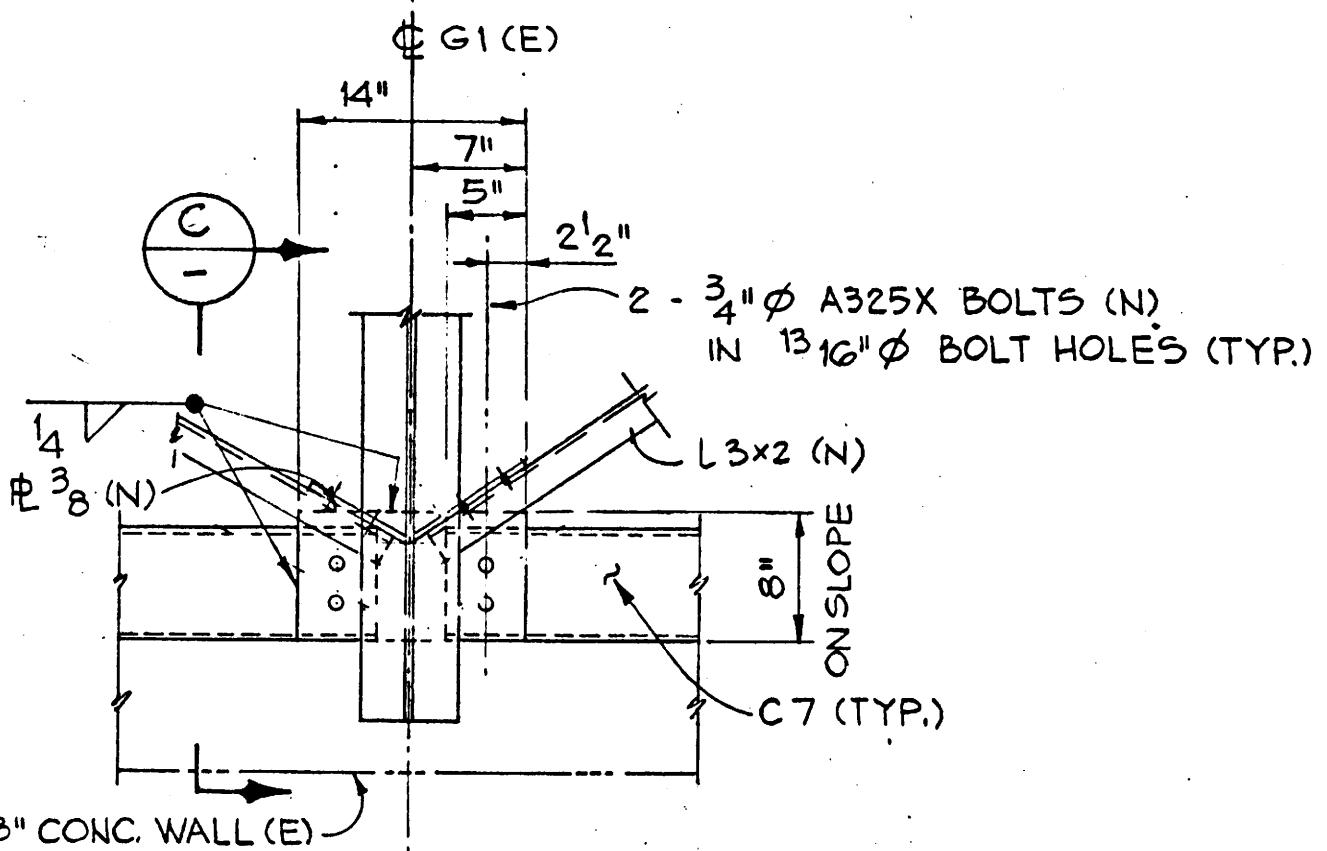
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FREDERIKSEN ENGINEERING			L.L.N.L. DISASTER CENTER BUILDING 313 SEISMIC REINFORCEMENT FUTURE WALL OPENINGS	DATE 7-16-81	DRAWN BY			
CONSULTING ENGINEERS ARCHITECTS			OAKLAND LONG BEACH	SCALE 1/2" = 1'-0"	CHK.			
				DRAWING NO.				
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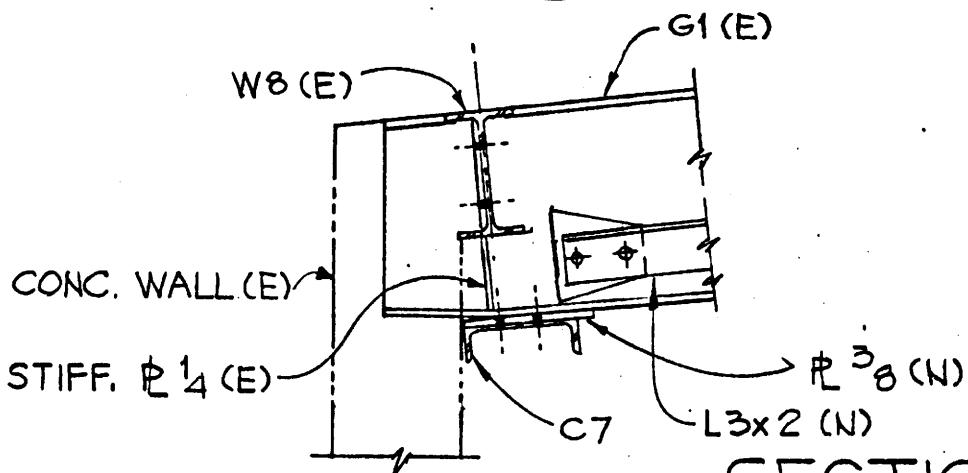


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CONSULTING ENGINEERS ARCHITECTS		OAKLAND LONG BEACH		SCALE 1" = 1'-0"		CHK. PCIN	
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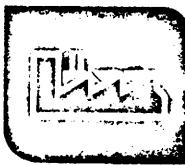
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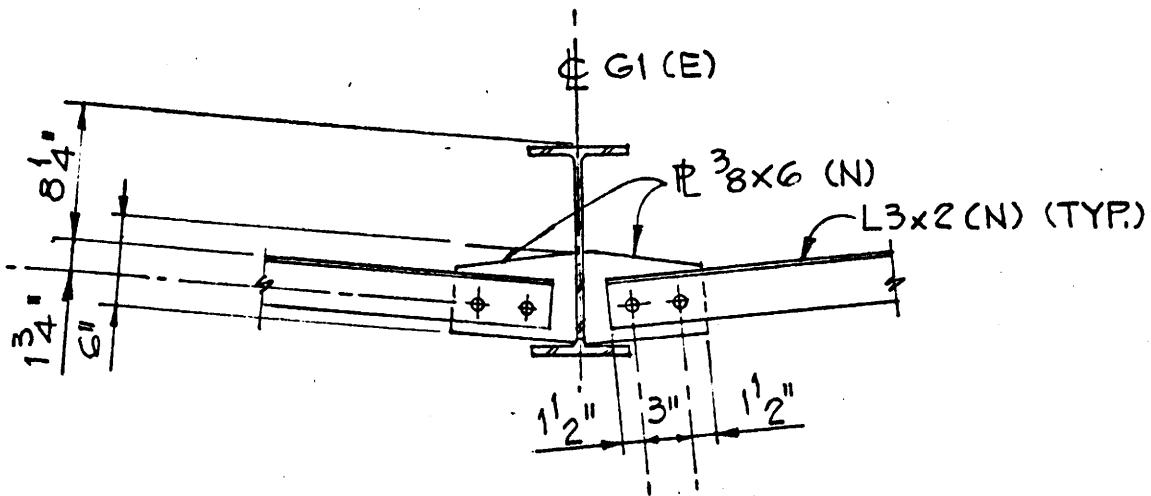
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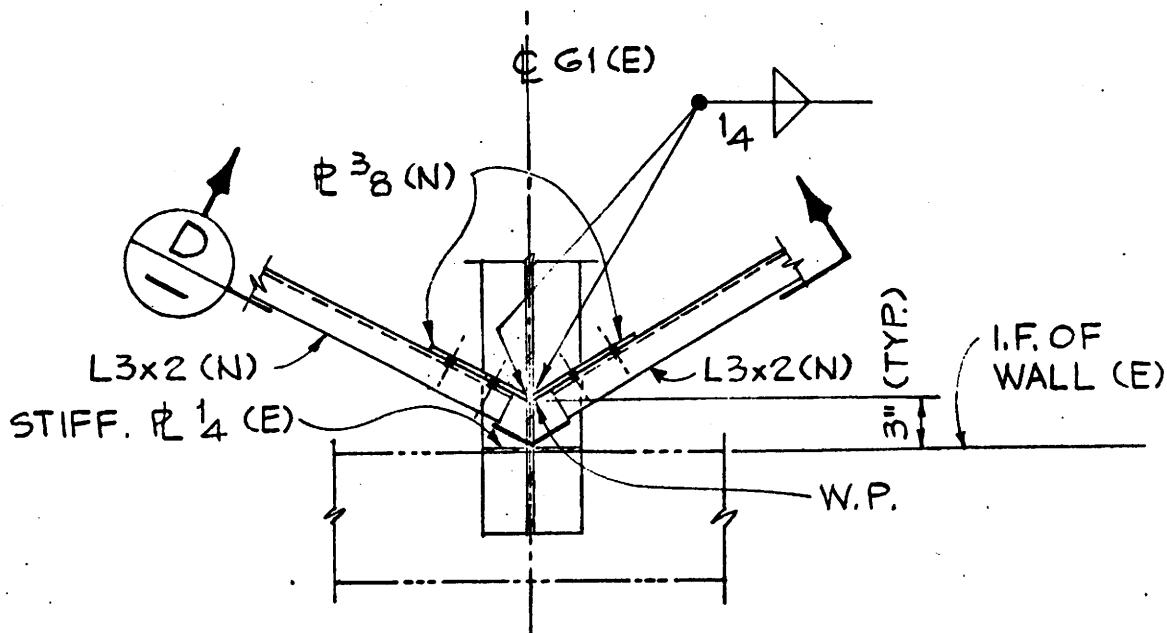
L.L.N.L. DISASTER CENTER
BUILDING 313
SEISMIC REINFORCEMENT
ROOF FRAMING SECTS.

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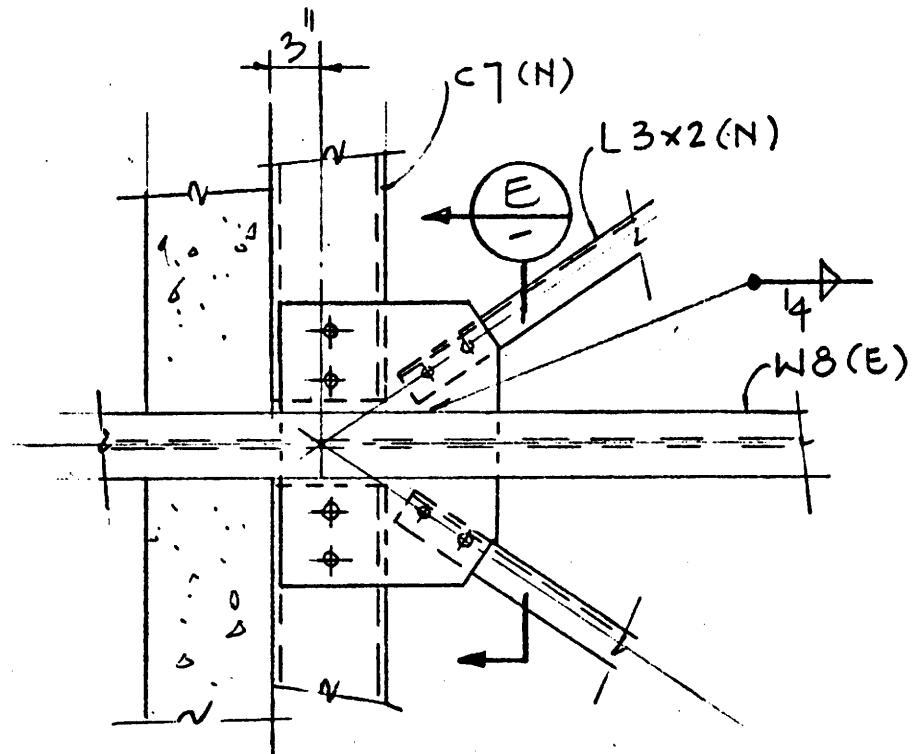
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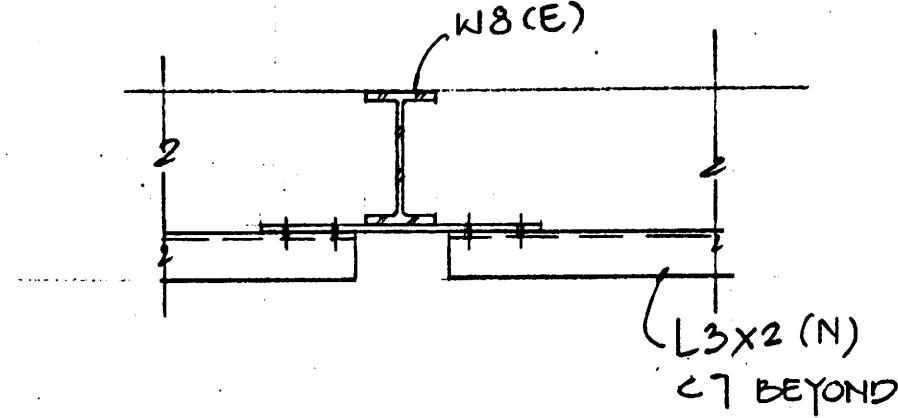
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FREDERIKSEN ENGINEERING CONSULTING ENGINEERS ARCHITECTS			L.L.N.L. DISASTER CENTER BUILDING 313 SEISMIC REINFORCEMENT ROOF FRAMING SECTS.	DATE 7-16-81 SCALE 1" = 1'.0		DRAWN BY	CHK. P/N	
			DRAWING NO.			REV.		
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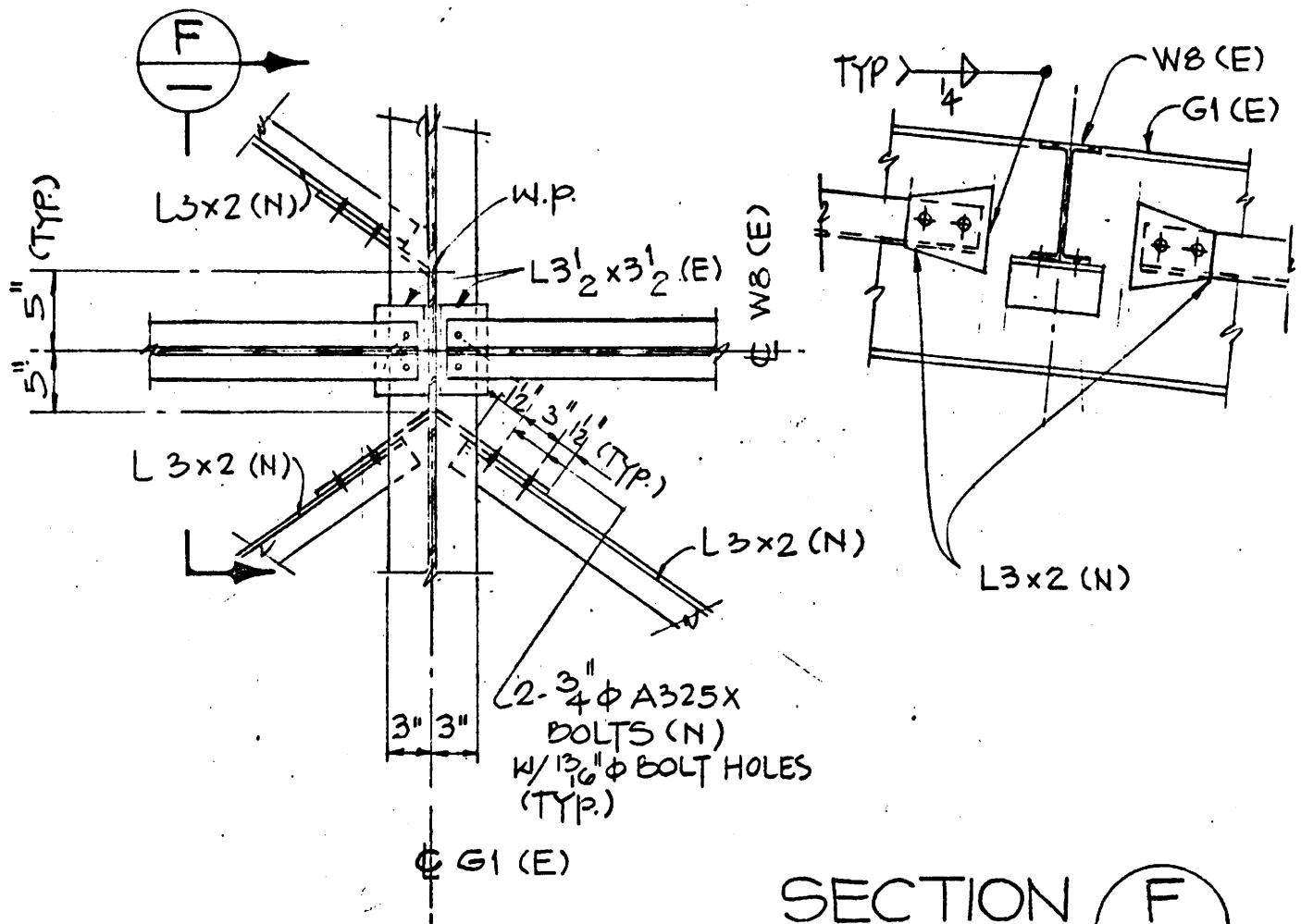
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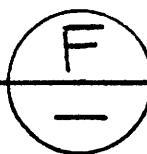
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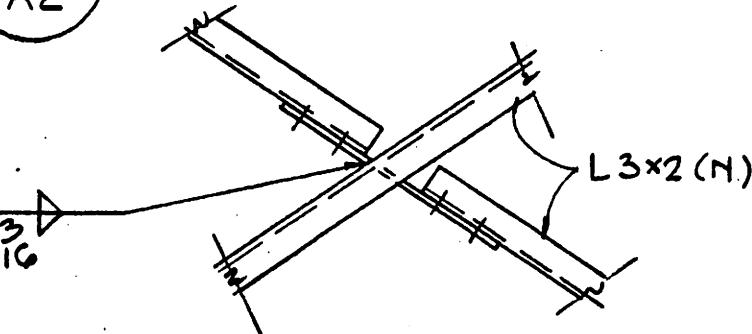
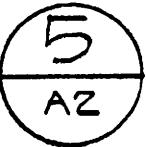
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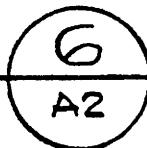
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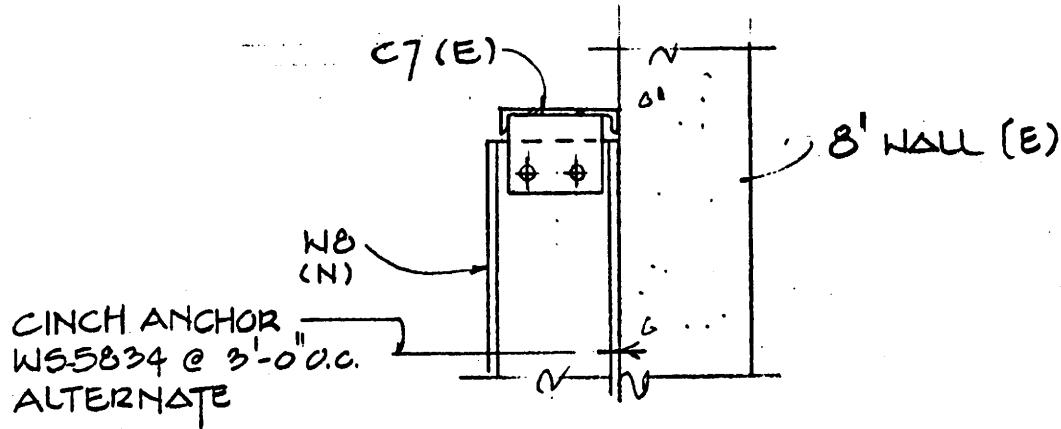
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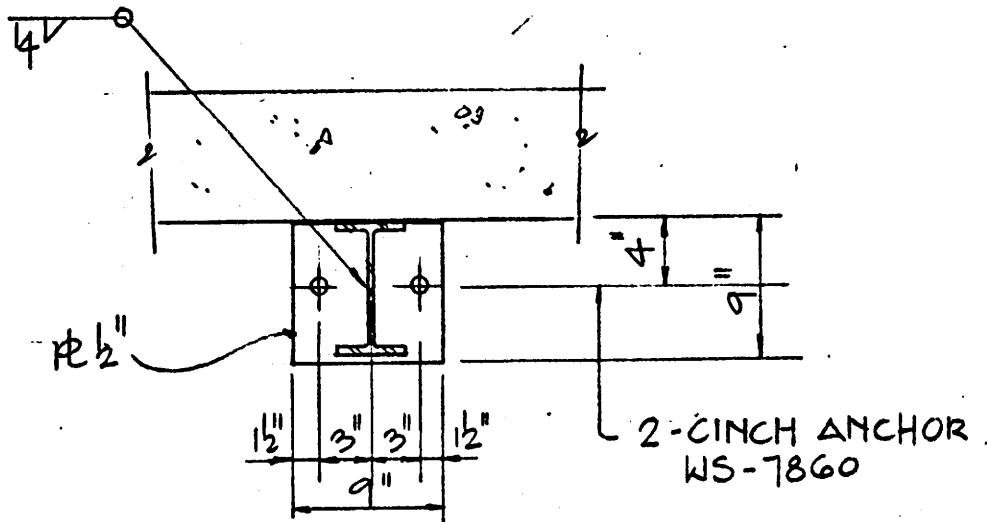
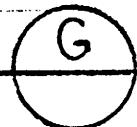
**FREDERIKSEN
ENGINEERING**
CONSULTING
ENGINEERS
ARCHITECTS
OAKLAND
LONG BEACH

L.L.I.L. DISASTER CENTER
BUILDING 313
SEISMIC REINFORCEMENT
ROOF FRAMING DET. & SECT.

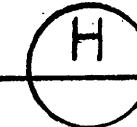
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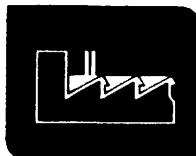
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FREDERIKSEN ENGINEERING CONSULTING ENGINEERS ARCHITECTS				L.L.N.L. DISASTER CENTER BUILDING 313 SEISMIC REINFORCEMENT FUTURE WALL OPNG. SECTS.	DATE 7-17-81 SCALE 1" = 1.0 DRAWING NO. 1354-A12	DRAWN 7-17-81 CHK. REV. 0	



**FREDERIKSEN
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OAKLAND

LONG BEACH

APPENDIX C

Wind-borne Missile Impact Test Report

Interdepartmental letterhead

Mail Station L- 322

Ext: 20780

February 12, 1981

TO: Distribution

FROM: B. R. Diaz

SUBJECT: Wind-Borne Missile Impact Tests - Summary of Test Results

The subject test series was conducted to establish the integrity of certain nonstructural building components of LLNL facilities to withstand wind-borne missile impacts of a specified design basis velocity. The missile for all tests was a 2 x 4 pine plank, 12 ft. long and weighing 15 lbs. Impact velocity for all tests was nominally 70 m.p.h. The full-scale targets, in chronological order, were: a 12 ft. x 14 ft. metal roll-up Hi-bay door; a metal, hollow-core, double pedestrian door; a metal, hollow-core, single pedestrian door; a chain link fence without wood slats; a chain link fence with wood slats; and four metal wall panel sections with gauge thickness of 22, 20, and 18.

The roll-up door, single and double pedestrian doors, and chain link fences all satisfactorily prevented entry or penetration of the 2 x 4 plank at 70 m.p.h. Four tests at 70 m.p.h. against wall panel sections opened up at the center lap joint in each test, allowing the 2 x 4 plank to penetrate about two feet. Two of the latter tests were conducted with the center joint of the panel fastened with sheet metal screws and two panels were tested with the center joint fastened with rivets. Neither the screws nor rivets exhibited satisfactory holding power for the level of impact loading experienced in these tests. In the case of the riveted 22 Gauge test, the panel material was weaker than the rivets which elongated the holes and punched through the panel material. In the case of the riveted 18 gauge test, the holes did not tear or elongate. The rivets failed in shear/tension. The panel tests concluded that the standard lap jointed sheet metal and rivet connections were not sufficient to prevent penetration of the missile.

Benjamin R. Diaz
B. R. Diaz
Civil Engineering Group Leader

BRD:sk



University of California

LAWRENCE LIVERMORE LABORATORY